Mechanical properties of large-scale parallel bamboo strand lumber

under local compression

Ke Zhou^{1a,1b}, Haitao Li^{1a,1b*}, Chaokun Hong^{1a,1b}, Dong Yang^{1a,1b}, Usama Sayed^{1a,1b}, Rodolfo Lorenzo², Ileana Corbi³, Ottavia Corbi³, Yingfeng Zuo⁴

² University College London, London WC1E 6BT, UK.

³ University of Naples Federico II, Via Claudio 21,80133 Naples, Italy.

Abstract: To study the local compression mechanical properties of large-scale parallel bamboo strand lumber (PBSL) structural elements, 20 specimens were tested considering two main influencing factors which are source position and compression direction. According to the experimental results, the failure of all specimens experienced elastic stage, elastic-plastic stage, and failure stage. The failure of the specimen was caused by the deformation perpendicular to the loading direction. For the middle specimens tangentially loaded, the failure belonged to ductile failure. For other group specimens, the failure belonged to quasi-brittle failure. The values for elastic modulus, stiffness, ultimate strength and Poisson's ratio of the end specimen were higher than those of the middle specimen. The values for elastic modulus, stiffness and Poisson's ratio of the specimens loaded tangentially were higher than those of the specimens loaded radially but the ultimate strength values for specimens loaded tangentially were lower than those for the specimens loaded radially. The ductility of the middle specimens was better than that of the end specimens. The load-displacement relationship model of the local compressive of PBSL was proposed based on the analysis. Based on the Ramberg-Osgood relation (ROR), the stress-strain relationship models for PBSL under local compression were proposed and compared with the test data.

Keywords: bamboo; local compression; damage mechanism; elastic modulus; ductility

1 Introduction

Similar to the timber [1-2], bamboo has a wide distribution, a great variety and excellent mechanical properties [3-6]. Besides, it takes a shorter harvest time for bamboo compared with trees. Therefore, the advantages of bamboo replacing wood for furniture, decoration and building structure are more and more obvious [7], which promotes the rapid development of the bamboo industry. The original bamboo materials couldn't meet the requirements of modern structure for the size and mechanical properties, promoting the development of engineered bamboo materials [8-9]. The common engineered bamboo materials are GluBam [10-12], parallel bamboo strand lumber (PBSL) [13], laminated bamboo lumber (LBL) [14-18]. Engineered bamboo materials could improve the shortcomings of original bamboo materials and could be used widely in the construction area.

Parallel bamboo strand lumber is made of processed bamboo fiber bundles [19]. With good mechanical performance, PBSL could meet the properties requirements of building structures. Ridzqo et al. [20] studied the fabrication of a novel composite board from bamboo fibers through a biologically binding mechanism by using fungal mycelium. Yu et al. [21] examined the effect of resin content and density on the properties of PBSL. Kumar et al. [22] investigated the effect of bamboo scrimber density on mechanical properties and water absorption properties. Li et al. [23-24] studied the basic mechanical properties of PBSL, as well as the compressive properties of PBSL considering different lengths and loading directions. Sharma et al. [25] examined the mechanical properties of PBSL and LBL and compared them with wood and wood products, which proved that the engineered bamboo has superior properties. Shangguan et al. [26-27] investigated the effect of different angles between load and grain and heat treatment on compressive properties of bamboo scriber and proposed the compressive strength model.

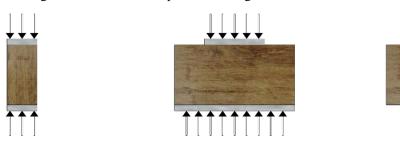
^{1a} College of Civil Engineering, Nanjing Forestry University, Nanjing 210037, China; ^{1b} Joint International Research Laboratory of Bio-composite Building Materials and Structures, Nanjing Forestry University, Nanjing 210037, China

⁴ College of Materials Science and Engineering, Central South University of Forestry and Technology, Changsha 410004, China *Corresponding author: Haitao LI, Professor, E-mail: lhaitao1982@126.com

Nugroho et al. [28-29] have performed a study to determine the suitability of the zephyr strand from Moso bamboo for structural composite board manufacture. Zhong et al. [30] studied the compressive strength of PBSL at different temperatures to provide the basis for its fire performance evaluation. Xu et al. [31] investigated the relationship between tensile and compressive stress-strain and failure mechanisms of PBSL at high temperatures. Malanit et al. [32] examined the physical and mechanical properties of bamboo scrimber made from an Asian bamboo. Zhou et al. [33-34] studied partial compression bearing capacity perpendicular and parallel to the grain of bamboo scrimber with small specimens, but the local compression properties of PBSL in two directions perpendicular to grain were not compared, and the research is only one side local compression and the size of the test specimen is not large enough to reflect the real situation. Besides the mechanical properties of materials, many scholars have also studied the structural components such as parallel bamboo strand lumber columns [35] and beams [36].

The engineered bamboo structure is similar to the wood structure. The local compression perpendicular to the grain of PBSL is a common stress state in the actual project, such as the connection between beams and columns, the connection between the main beam and the secondary beam, the use of PBSL as wood sleeper, cribbing, dunnage so on. At present, there have been some types of research on the local compression perpendicular to the grain of the timber. Van der Put [37] has given the theoretical explanation of the bearing strengths of locally loaded timber blocks, based on the equilibrium method of plasticity. Leijten et al. [38-40] have investigated the test methods for the local compression perpendicular to grain properties of wood perpendicular to the grain and compared with different national design codes. Madsen et al. [41] emphasized the influence of geometric conditions on the partial compression bearing capacity analyzed the stress diffusion by numerical simulation.

Yeh and Lin [42] investigated the effect of growth height on compressive strength for laminated bamboo lumber specimens with a cross-section of 30mm × 30mm. The compressive strengths in the lower growth height were smaller than those values in the upper growth height both for jointed and un-jointed specimens. Yeh and Lin [42] have shown a significant dependence of compressive strength on the growth height of the source bamboo. Their observation was made using small scale laminate bamboo specimens. This study will also investigate how the source bamboo growth height influences the compressive strength of full-scale PBSL structural members.



(a) Full section loading (b) Local compression on one side (c) Local compression on both sides

Fig. 1. Different compression situations

For the three compression situations as shown in Fig. 1. The situation shown in Fig.1a and Fig.1b has been studied, but it is still blank for the situation shown in Fig.1c. Thus, for the situation of Fig.1c, local compression tests about large-scale PBSL specimens were carried out in this paper considering the influencing factors of different parts and compression direction. Both the local stress-strain behaviour and the global load-displacement behaviour of the large PBSL specimens will be investigated. This is necessary because the global load-displacement is effected by splitting and delamination within the structural member, while the local stress-strain behaviour demonstrates the inelastic behaviour of the material itself. Detailed analysis and discussion about the test results have been done.

2 Materials and Test Methods

Harvested at the age of 3–4 years, Moso bamboo (*Phyllostachys pubescent*) was chosen to manufacture the specimens. As can be seen from Fig. 1, original bamboo tubes (Fig.2a) were split into 20 mm wide strips (Fig.2b) with the outer skin (epidermal) and inner cavity layer (pith peripheral) removed, and all culm strips were split into bamboo filament bundles (Fig.2c). All bundles were immersed in the resin pool and then pulled out for drying. Then

the bundles were put into the molds (Fig.2d) and pressed together to form the blocks (Fig.2e) under 20°C with the press pressure about 62 MPa to achieve the aim density value of 1200 kg/m³. The molds with the blocks were then cured with the temperature about 135°C for 9 hours. The final moisture content was 8.22%. Phenol glue was used to produce the PBSL specimens. The percentage of adhesive in the final composite material is 12%.

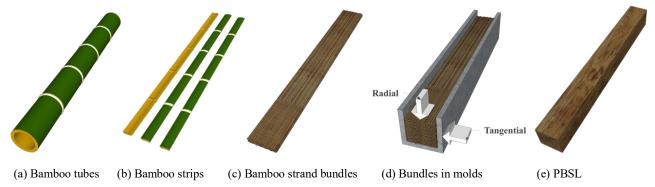


Fig. 2. The production process of PBSL.

Twenty specimens were cut from the PBSL, with a length of 200 mm and a section of 100 mm× 100 mm. The truncation position is in the middle and the end part. Bamboo strand bundles in the end part are the upper position of the original bamboo. The specimens were divided into four groups based on different parts and different loading directions and each group contains 5 identical specimens. The average density of the end specimens is 1.21g/cm³ and the middle specimens are 1.24g/cm³. Fig. 3. shows the specimens from two main parts and loading directions. Detailed information about the specimens could be seen in Table 1.

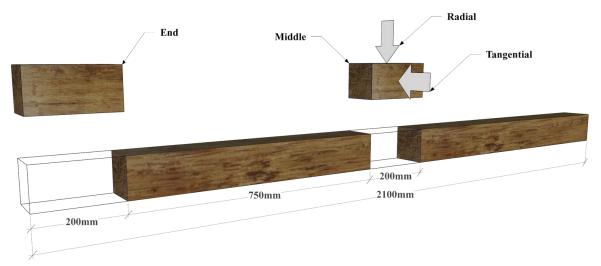


Fig. 3. Diagram of different parts and loading directions

Table 1 The information for the specimens

Group	Material location	Loading direction	Width(mm)	Height(mm)	Length(mm)
ET	End	Tangential	100	100	200
ER	End	Radial	100	100	200
MT	Middle	Tangential	100	100	200
MR	Middle	Radial	100	100	200

Notes: ET represents the specimen cut from the end part of the original PBSL blocks and loaded tangentially; ER represents the specimen cut from the end part of the original PBSL blocks and loaded radially; MT represents the specimen cut from the middle part of the original PBSL blocks and loaded tangentially; MR represents the specimen cut from the middle part of the original PBSL blocks and loaded radially.

The local bearing area of the test specimen is $100 \text{ mm} \times 100 \text{ mm}$, and $100 \text{ mm} \times 150 \text{ mm} \times 30 \text{ mm}$ stainless-steel blocks were set up on the top and bottom of the specimen. By loading on stainless-steel blocks to achieve the

purpose of local loading compression, as shown in Fig. 5. The displacement was measured between the opposing faces of the loading plates by using laser displacement sensors.

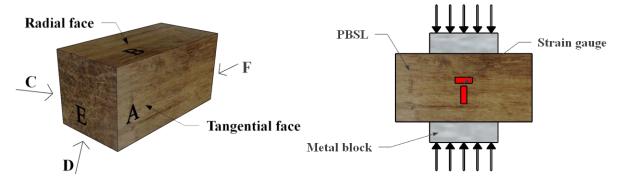


Fig. 4. Schematic diagram of specimens

Fig. 5. Loading method

The test was performed using a microcomputer-controlled electro-hydraulic servo universal testing machines with a capacity of 2000 kN and a TDS Data Acquisition System. The total time is controlled within 8 to 10 minutes from the start of loading and the speed of loading is 2 mm/min. To describe the experimental phenomenon conveniently, all faces of the specimens are numbered, and the four sides are A, B, C, D and two cross-sections are E, F; the pressurized faces are B and D. Specimen radially loaded is shown in the fig. 3. For specimens tangentially loaded, tangential faces are marked as B and D and radial faces are marked as A and C.

3 Failure Phenomena Analysis

The load-displacement curves obtained from the tests are shown in Fig. 6. The failure of the specimen could be divided into three stages: elastic stage, elastic-plastic stage, and failure stage. When a load value of all specimens increased to about 40% of the ultimate load, the specimens began to enter into the elastic-plastic stage. At the end of the elastic-plastic stage, about 90% of the ultimate load, an audible noise could be heard before visual cracks appeared on the surface. With the development of cracks, the specimens gradually broke down. The failure model of the specimens under the same loading direction was similar. The failure of the specimens was mainly caused by the deformation perpendicular to the loading direction.

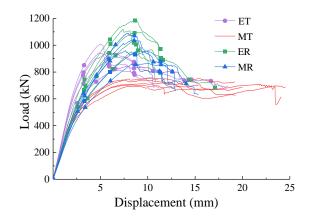


Fig. 6. Load-displacement curves

3.1 The failure of specimens tangentially loaded

There are two main failure models for specimens tangentially loaded, both of which belong to strength failure. The failure model I is shown in Fig. 7. At the initial stage of loading, the specimens were in the elastic stage, and the vertical strain for face A and C were equal to each other without crack on the specimens. As the specimen entered into the elastic-plastic stage, plastic deformation occurred at the contact edge between the specimen and the upper and lower stainless-steel blocks while there was no obvious change on the surface of the specimens. When the load

increased to about 90% of the ultimate load, cracks began to appear on the surface of the specimen. The horizontal cracks appeared firstly near the apex corner of face B and D, as shown in Fig. 7. (c) and (f). With the increase of load, the cracks developed, and the vertical cracks appeared near the apex corner of cross-section E and F. The vertical cracks developed to the middle of the specimen, and finally became two main vertical bending cracks. The cracks developed towards the middle and the distance of the two main cracks became smaller as shown in Fig. 7. (a) and (d). It can be seen from the damaged specimen that the deformation of the compression area in the middle of the specimen was obvious. Several main cracks could be seen from the side parts for face B and D, as shown in Fig. 7. (c) and (f), and there is no obvious failure on face A and C of the specimen, as shown in Fig. 7. (b) and (e).

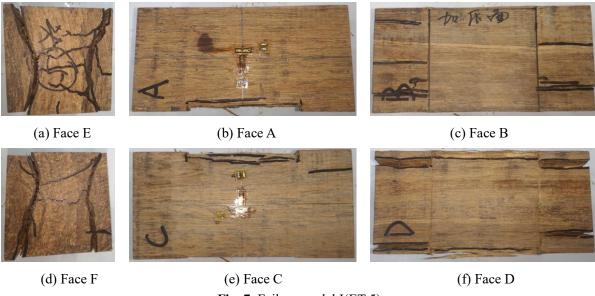


Fig. 7. Failure model I(ET-5)

The failure model II is similar to model I, but the vertical deformation of the specimen for failure model II is more obvious than that for failure model I as shown in Fig. 8. (b) and (e). There are more horizontal cracks on the compression faces B and D as shown in Fig. 8. (c) and (f). There was one more vertical main crack in the middle of cross-section E and F for failure model II, and the failure of the specimen was mainly caused by these three cracks, as shown in Fig. 8. (a) and (d).

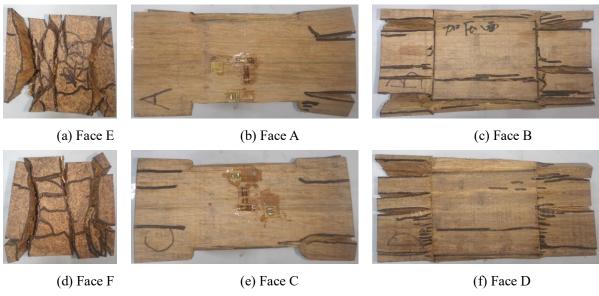


Fig. 8. Failure model II (MT-4)

The crack developed inside of the specimen. The failure of specimens tangentially loaded was mainly caused by the horizontal deformation perpendicular to the loading direction and grain direction as shown in Fig. 9.

As shown in table 2, most of the specimens from group ET belong to failure model I while most of the

specimens from group MT belong to failure model II. Bamboo strand bundles for group ET were from the upper position of the original bamboo compared with those for group MT. As the former studies have discovered mechanical differences between small specimens sourced from different growth portions, the differences also existed in the large-scale parallel bamboo strand lumber specimens. Due to the high density of bamboo fibers for the bundles from the upper position, the strength for group ET are stronger than that for group MT and the main failure models were also different for two groups.

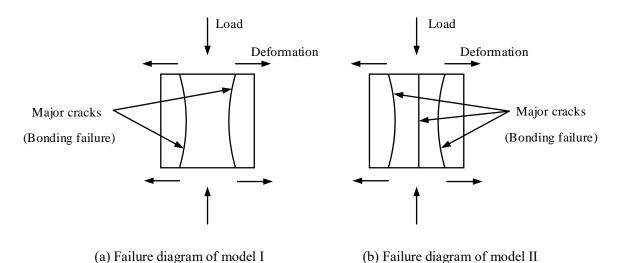


Fig. 9. Deformation leading to failure

3.2 The failure of specimens radially loaded

Two main failure models of specimens radially loaded were also strength failure. The failure model III is shown in Fig. 10. At the initial stage of loading, it is the same as the specimens tangentially loaded. When the load increased to about 90% of the ultimate load, the specimen began to damage. The vertical cracks first appeared in the middle of face A and C and the apex corner of sections E and F. The crack of sections E and F finally showed a 'V' shape as shown in Fig. 10.

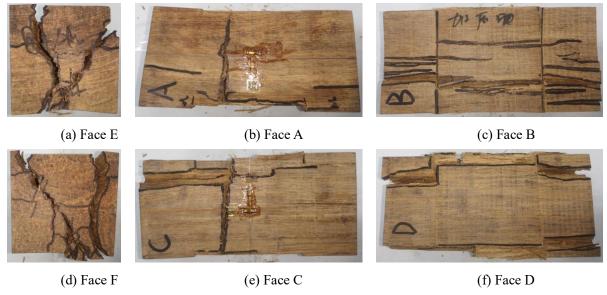


Fig. 10. Failure model III (ER-4)

The cross-section of the failure model IV was similar to the failure of specimens tangentially loaded as shown in Fig. 11. There were two main vertical cracks on face E and F. Similar cracks appeared on face A and C as failure model III. The failure of the specimen was caused by the falling off and fracture of face A and C.

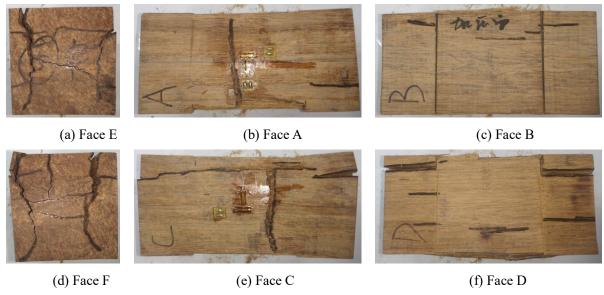
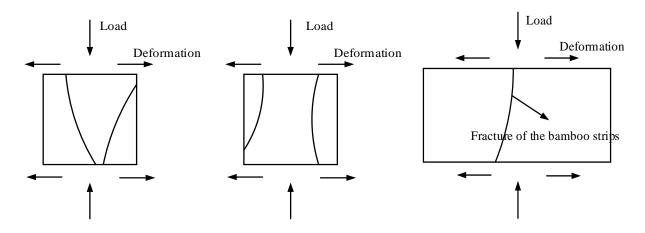


Fig. 11. Failure model IV (MR-5)

The two failure of the specimen radially loaded have obvious signs, and the degree of failure is more serious than the specimen tangentially loaded. The failure was caused by the development of vertical cracks on cross-section E, F and face A and C, which leaded to the partial falling off and fracture of faces A and C. The two failure models were not only caused by the horizontal deformation perpendicular to the loading direction and bamboo fiber direction but also the horizontal deformation parallel to bamboo fiber direction shown in Fig. 12.



(a) Failure diagram of model III (b) Failure diagram of model IV (c) Failure diagram of both models Fig. 12. Deformation leading to failure

Comparing the main failure models of specimens loaded by different directions, the failure of the specimens tangentially loaded was mainly due to the bonding failure between the bamboo fibers, which caused the face A and C to partly fall off. The bamboo fibers of the specimens tangentially loaded did not break in the length direction.

The failure of the specimen radially loaded was not only the bonding failure between the bamboo fibers but also the fracture of bamboo fibers which could be seen from the vertical cracks on face A and C, so the failure degree of specimens radially loaded is deeper.

4 Combined Analysis

4.1 Data comparison and analysis

Table 2 shows the experimental results.

Table 2 The experimental results of specimens

Specimen	$\rho(g/cm^3)$	E(MPa)	k(kN/mm)	f(MPa)	v	$s_{\rm cu}$ (mm)	$s_{\rm cy}({\rm mm})$	μ	Failure model
ET-1	1.23	6083	313	100.8	0.24	5.13	2.08	2.46	I
ET-2	1.25	5103	298	92.5	0.26	6.70	1.51	4.43	I
ET-3	1.20	5260	316	94.4	0.24	5.08	1.65	3.07	II
ET-4	1.22	5472	295	94.7	0.24	5.83	1.73	3.36	I
ET-5	1.23	5860	277	84.0	0.25	6.79	1.56	4.36	I
Mean	1.23	5556	300	93.3	0.25	5.91	1.71	3.54	1
SDV	0.0162	366	14.02	5.41	0.0080	0.735	0.202	0.760	
COV	0.0102	0.0659	0.0468	0.0580	0.0000	0.735	0.202	0.215	
CHV	1.19	4818	272	82.4	0.0323	4.42	1.30	2.01	
MT-1	1.16	4396	251	73.9	0.24	7.45	1.35	5.54	II
MT-2	1.15	4117	250	74.8	0.23	8.05	1.68	4.79	II
MT-3	1.20	4026	240	69.3	0.23	8.11	1.91	4.25	I
MT-4	1.23	3765	251	70.0	0.25	7.06	1.66	4.24	II
MT-5	1.22	3631	245	70.3	0.25	7.11	1.51	4.72	II
Mean	1.19	3987	247	71.7	0.24	7.56	1.62	4.71	п
SDV	0.0319	269	4.32	2.24	0.0089	0.449	0.187	0.475	
COV	0.0267	0.0675	0.0175	0.0312	0.0373	0.0594	0.115	0.101	
CHV	1.13	3445	239	67.1	0.22	6.65	1.25	3.75	
ER-1	1.23	3726	225	96.4	0.22	8.28	2.32	3.57	III
ER-2	1.32	4253	235	118.6	0.23	8.63	2.15	4.02	IV
ER-3	1.27	3781	238	112.0	0.20	7.39	2.30	3.22	III
ER-4	1.23	3650	231	109.6	0.20	8.34	2.41	3.46	III
ER-5	1.22	3610	230	102.9	0.20	8.46	2.01	4.20	IV
Mean	1.25	3804	232	107.9	0.21	8.22	2.24	3.69	
SDV	0.0372	232	4.45	7.63	0.0126	0.432	0.141	0.363	
COV	0.0297	0.0610	0.0192	0.0708	0.0602	0.0525	0.0632	0.0981	
CHV	1.18	3336	223	92.5	0.18	7.35	1.95	2.96	
MR-1	1.25	3504	211	108.9	0.19	7.97	1.99	4.01	IV
MR-2	1.27	3366	217	104.0	0.19	8.42	1.76	4.78	III
MR-3	1.20	2982	212	94.3	0.20	9.67	1.87	5.18	III
MR-4	1.20	3194	198	85.8	0.21	8.28	2.02	4.10	IV
MR-5	1.20	3180	224	95.9	0.17	7.67	1.88	4.08	IV
Mean	1.22	3245	212	97.8	0.19	8.40	1.90	4.43	
SDV	0.0301	178	8.55	8.02	0.0133	0.685	0.0931	0.468	
COV	0.0246	0.0547	0.0402	0.0820	0.0691	0.0815	0.0489	0.106	
CHV	1.16	2887	195	81.6	0.17	7.02	1.72	3.49	
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Notes: In Table 1, ρ is density; E is elastic modulus (Select the point calculation when the strain is 0.005.); k is stiffness (Select the point calculation when the axial displacement is 1.5mm.); f is local bearing ultimate strength perpendicular to grain; v is Poisson's ratio (It's the Poisson's ratio in the elastic stage.), s_{cu} is the ultimate displacement, s_{cy} is the displacement for the yield point (The yield point is the point at which the stiffness begins to decrease. As shown in Fig 13, When the difference is between 0 and 20kN, the corresponding point can be selected as the yield point.), μ is ductility coefficient (The displacement corresponding to the ultimate load divided by the displacement corresponding to the yield point, s_{cu}/s_{cy} , SDV is the standard deviation; COV is the coefficient of variation (SDV/Mean); CHV means characteristic value, calculated on the basis that 95% of samples exceed the characteristic value (Mean-2.015 × SDV).

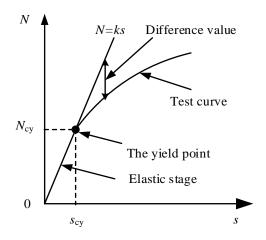


Fig. 13. Determination of yield point

From the results in table 1, the coefficient of variation of elastic modulus, stiffness, ultimate strength and Poisson's ratio were all less than 0.1, and the data are stable. The relationship between the elastic modulus, stiffness, ultimate strength and Poisson's ratio of each group of specimens can be expressed by formula (1) to formula (4):

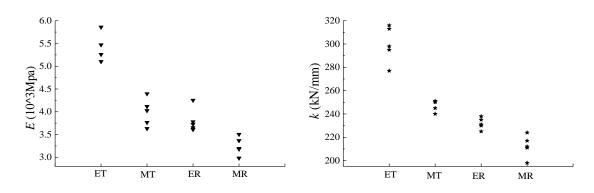
$$E_{\rm ET} = 1.393 E_{\rm MT} = 1.460 E_{\rm ER} = 1.712 E_{\rm MR} \tag{1}$$

$$k_{\rm ET} = 1.137 k_{\rm MT} = 1.219 k_{\rm ER} = 1.391 k_{\rm MR}$$
 (2)

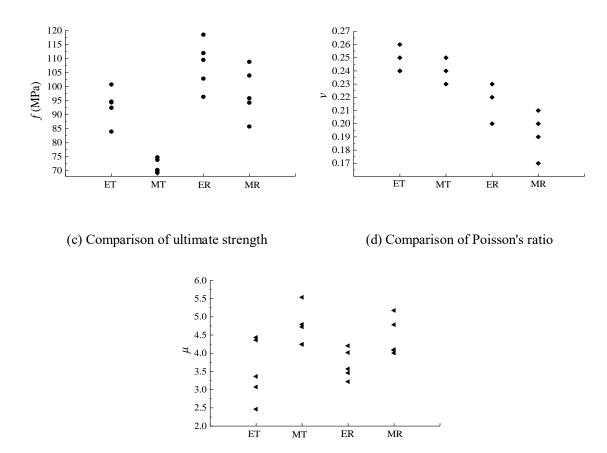
$$f_{\rm ET} = 1.227 f_{\rm MT} = 0.890 f_{\rm ER} = 1.009 f_{\rm MR}$$
 (3)

$$v_{\rm ET} = 1.045 v_{\rm MT} = 1.277 v_{\rm ER} = 1.353 v_{\rm MR}$$
 (4)

Fig.14. Compared the elastic modulus, rigidity, ultimate strength, Poisson's ratio and ductility coefficient of each group of specimens.



- (a) Comparison of elastic modulus
- (b) Comparison of stiffness



(e) Comparison of ductility coefficient

Fig.14. Comparison of experimental results

It could be found that for ultimate strength, elastic modulus, Poisson's ratio, stiffness, the end specimens were higher than the middle specimens. The elastic modulus, Poisson's ratio and stiffness of the specimens tangentially loaded were higher than that for specimens radially loaded, and the ultimate strength was lower than that radially loaded. The ductility of the middle specimens was better than that of the end specimens. There was little difference in the ductility of specimens loaded in different directions.

4.2 Load-displacement curves

The load-displacement curves obtained from the tests are shown in Fig. 6. All specimens went through the elastic stage, elastic-plastic stage, and failure stage. In the elastic stage, the load-displacement curves were straight, and the rigidity was stable. The stiffness of the specimen decreases during the elastic-plastic phase, and the curve occurred deviation. In the failure stage, the bearing capacity of the middle specimen tangentially loaded did not decrease and the ultimate load remained unchanged. The bearing capacity of the other three groups of specimens decreased in the failure stage. The stiffness of the four groups of specimens in the elastic stage slightly deviated. When the vertical displacement of the specimens reached 1.5-2.5mm, the specimens entered the elastic-plastic stage, and when the vertical displacement reached 5-9mm, the specimens entered the failure stage.

Combined with the failure model of the specimens, the load-displacement relationship of the four groups of specimens could be expressed by a function model as shown in Fig. 15.

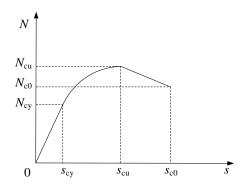


Fig. 15. Load-displacement model

The load-displacement model could be expressed by formula (5):

$$N = \begin{cases} ks & (0 \le s < s_{cy}) \\ \lambda_1 s^2 + \lambda_2 s + \lambda_3 & (s_{cy} \le s < s_{cu}) \\ N_{cu} - \alpha k (s - s_{cu}) & (s_{cu} \le s < s_{c0}) \end{cases}$$
 (5)

Where $s_{\rm cy}$ is the displacement for the yield point; $N_{\rm cy}$ is the load for the yield point; $s_{\rm cu}$ is the ultimate displacement; $N_{\rm cu}$ is the ultimate load; $s_{\rm c0}$ is maximum displacement when the specimen is broken; $N_{\rm c0}$ is the load corresponding to $s_{\rm c0}$; k is stiffness; k, λ_1 , λ_2 , λ_3 , α could be solved by formula (6) to formula (10):

$$k = \frac{N_{\rm cy}}{s_{\rm cy}} \tag{6}$$

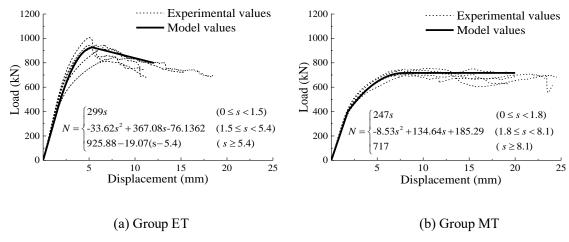
$$\lambda_{1} = \frac{N_{\text{cy}} - N_{\text{cu}}}{(s_{\text{cy}} - s_{\text{cu}})^{2}} \tag{7}$$

$$\lambda_2 = \frac{2s_{\text{cu}}(N_{\text{cy}} - N_{\text{cu}})}{(s_{\text{cy}} - s_{\text{cu}})^2}$$
 (8)

$$\lambda_3 = N_{\rm cy} - \lambda_1 s_{\rm cy}^2 - \lambda_2 s_{\rm cy} \tag{9}$$

$$\alpha = \frac{N_{c0} - N_{cu}}{k(s_{c0} - s_{cu})} \tag{10}$$

The comparison between the test results of each group of specimens and the theoretical values of the model as shown in Fig. 16. The model can represent the load-displacement relationship.



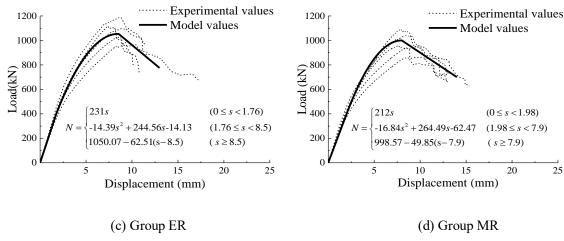


Fig. 16. Comparison between experimental values and model values

From the load-displacement curves and models of four groups of specimens, it could be seen that in the elastic stage, the stiffness of the end specimen was greater than that of the middle specimen, and that of the specimen tangentially loaded was greater than that of the specimen radially loaded. The ultimate load of the end specimen was greater than that of the middle specimen, and that of the specimen radially loaded was greater than that of the specimen tangentially loaded.

The falling speed of the load in the failure stage reflects the failure degree of the specimen. From the load-displacement model, it could be seen that the failure of the end specimen was more serious than that of the middle specimen, and the failure of the specimen loaded radially was more serious than that of the specimen loaded tangentially. It was consistent with the failure models. For the middle specimen tangentially loaded, the failure degree was least, and the specimen was compacted during the loading process so that the bearing capacity of this group of specimens in the failure stage could remain unchanged within a certain deformation range.

4.3 Stress-strain curves

The stress-strain curves obtained from the tests were shown in Fig. 17. The stress-strain curves were similar to the load-displacement curves in the elastic stage and the elastic-plastic stage. As the cracks appeared in the specimen at the end of the elastic-plastic stage, the internal force was redistributed. When the cracks appeared and developed in the specimen, most of the strain gauges stopped working or the measured values couldn't reflect the true strain of the specimen. Therefore, the stress-strain curves had no obvious decline stage. It could be found that the vertical ultimate strain of the specimens loaded radially was larger than that of the specimens loaded tangentially.

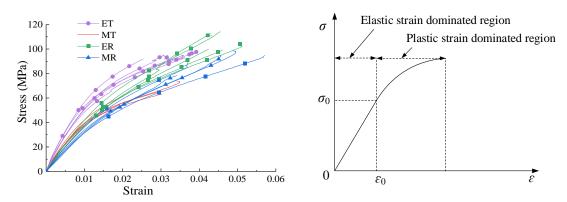


Fig. 17. Stress-strain curves

Fig. 18. Ramberg-Osgood Model

It could be seen from Fig. 17. that there was no obvious yield point in the stress-strain curves. Ramberg-Osgood relation (ROR) could be used to express the stress-strain relationship [43], as shown in Fig. 18. In the original form of the ROR model, the total strain could be expressed as the sum of elastic strain (ε_e) and plastic strain (ε_p).

$$\varepsilon = \frac{\sigma}{E} + K(\frac{\sigma}{E})^n \tag{11}$$

By substituting the stress (σ_0) and strain (ε_0) at any point in the elastic stage into the formula (11), the elastic modulus (E) could be eliminated and the ROR model could be obtained in formula (12).

$$\frac{\mathcal{E}}{\mathcal{E}_0} = \frac{\sigma}{\sigma_0} + \alpha \left(\frac{\sigma}{\sigma_0}\right)^n \tag{12}$$

Where α and n are undetermined parameters which could be determined by fitting experimental values.

The stress-strain relationship of each group of specimens obtained by nonlinear fitting was shown in Fig. 19. ROR model could better express the stress-strain relationship of PBSL under local compression.

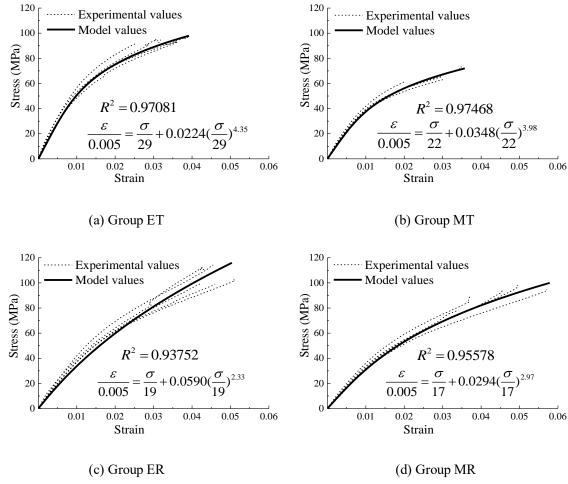


Fig. 19. Comparison between experimental values and model values

5 Conclusions

To evaluate how the cutting position and compression direction influence the local compression mechanical properties of large-scale parallel bamboo strand lumber (PBSL), 20 specimens were tested and the experimental results were analyzed. The following conclusions are drawn.

- (1) The failure of all specimens has gone through the elastic stage, elastic-plastic stage, and failure stage. For the middle specimens tangentially loaded, the failure belonged to ductile failure. For other group specimens, the failure belonged to quasi-brittle failure. The failure of the specimen loaded radially was more serious than that of the specimen loaded tangentially.
- (2) The failure of the specimen was caused by the deformation perpendicular to the loading direction. The failure of the specimen tangentially loaded was mainly caused by deformation perpendicular to grain direction, while the specimen radially loaded was not only caused by the deformation perpendicular to

- grain direction but also the deformation parallel to bamboo fiber direction grain direction.
- (3) The values for elastic modulus, stiffness, ultimate strength and Poisson's ratio of the end specimen were higher than those of the middle specimen. The values for elastic modulus, stiffness and Poisson's ratio of the specimens loaded tangentially were higher than those of the specimens loaded radially but the ultimate strength values for specimens loaded tangentially were lower than those for the specimens loaded radially. And the ductility of the middle specimens was better than that of the end specimens.
- (4) The load-displacement relationship model of the local compressive of PBSL was proposed according to the analysis of the test data. Based on the Ramberg-Osgood relation (ROR), the stress-strain relationship models for PBSL under local compression were proposed and compared with the test data.

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