

1 **Enhanced Composite Plate Impact Damage Detection and Characterisation using**
2 **X-Ray Refraction and Scattering Contrast Combined with Ultrasonic Imaging**

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10

11 **Abstract**

12 Ultrasonic imaging and radiography are widely used in the aerospace industry for non-
13 destructive evaluation of damage in fibre-reinforced composites. Novel phase-based X-
14 ray imaging methods use phase effects occurring in inhomogeneous specimens to
15 extract additional information and achieve improved contrast. Edge Illumination
16 employs a coded aperture system to extract refraction and scattering driven signals in
17 addition to conventional absorption. Comparison with ultrasonic immersion C-scan
18 imaging and with a commercial X-ray CT system for impact damage analysis in a small
19 cross-ply carbon fibre-reinforced plate sample was performed to evaluate the potential
20 of this new technique. The retrieved refraction and scattering signals provide
21 complementary information, revealing previously unavailable insight on the damage

22 extent and scale, not observed in the conventional X-ray absorption and ultrasonic
23 imaging, allowing improved damage characterisation.

24 **Keywords:** B. Impact behaviour; D. Non-destructive testing; D. Radiography; D.
25 Ultrasonics

26

27 **1. Introduction**

28 Carbon fibre-reinforced composites are widely used in the aerospace industry as
29 they offer high strength to weight ratios [1]. However, carbon fibres are brittle, and
30 carbon fibre-reinforced composites are prone to Barely Visible Impact Damage (BVID)
31 [2]. Matrix cracking is typically the initial type of BVID to occur and thus the most
32 common for transverse impacts [3]. Different types of crack can form (e.g., shear,
33 bending cracks), depending on the stresses involved. Delaminations are initiated by
34 transverse impact and are the most severe type of BVID [2]. Such defects affect the
35 structural integrity of the composites, potentially leading to complete failure. The
36 detection and identification of such defects is important for predicting the composite
37 structure health and performance capabilities. In the aerospace industry, carbon fibre-
38 reinforced polymers (CFRP) have a permitted 0.4% compressive strain to failure, and as
39 a result the cost of maintenance and inspection of the parts is high [4].

40 Different Non-Destructive Evaluation (NDE) techniques can be used for the
41 monitoring of composites, including ultrasonic imaging, radiography, eddy current
42 testing, magnetic and thermographic testing [5-7]. Fast, accurate and cost-effective
43 techniques are required, with radiographic CT and ultrasonic imaging the main

44 techniques used today in the aerospace industry [8]. Immersion ultrasonic testing allows
45 for automated scanning of a specimen with uniform coupling, and thus sensitivity, as
46 well as the possibility of scanning irregular shapes and surfaces [8]. Immersion ultrasonic
47 C-scans are used for the detection of defects and delaminations in the normal plane,
48 whereas cracks oriented parallel to the emitted waves are unlikely to be observed [9].
49 Automated ultrasonic NDE systems for the detection of manufacturing defects in
50 composites such as voids, irregular fibre volume fraction, ply stacking sequence, fibre
51 waviness, and out-of-plane fibre wrinkling were developed [10, 11]. The main
52 disadvantage of ultrasonic imaging is its resolution limitation, as the signal attenuation
53 increases with increasing frequency, thus requiring a trade-off between resolution and
54 inspection depth. Small features such as micro-defects or individual fibres cannot be
55 resolved in composite plates using ultrasonic imaging. Another disadvantage of
56 ultrasonic C-scan imaging is its limitation for the detection of multiple defects across the
57 thickness of the sample, as a large fraction of the signal reflects or scatters at the first
58 defect [9].

59 In conventional X-ray imaging, the absorption rate of X-rays depends on the
60 attenuation coefficient and thickness of the features studied, hence large enough
61 features present in a composite plate can be observed as variations in the detected
62 intensity due to the fraction of X-rays absorbed [8]. X-ray computed tomography (CT) is
63 the most commonly used imaging technique for the detection of defects in composite
64 plates, such as delaminations, porosity and cracks, which are not visible in 2D images
65 [12]. Ultrasonic C-scans are often used to prove the viability of new CT reconstruction
66 methods or experimental setups [13]. Comparisons involving radiographic imaging and
67 ultrasonic imaging techniques were performed to study the evolution of damage

68 (cracks, delaminations) in self-healing composites [14], for the detection of BVID in
69 composites (e.g. matrix and fibre cracks, debonding and fibre pullout) [15], as well as
70 the evaluation of porosity content in composites plates, which is more difficult to
71 estimate using ultrasonic imaging alone [16]. Both imaging methods are capable of
72 detecting damage in fibre-reinforced composite plates, with X-ray CT imaging offering
73 higher resolutions and capable of detecting micro-damage and individual plies, up to the
74 detection of individual fibres [13–15].

75 For defects in composite materials too small to cause sufficient variation in the
76 detected intensity and invisible using conventional radiography, X-ray Phase Contrast
77 imaging (XPCi) offers an advantageous alternative. XPCi is different from conventional
78 X-ray imaging as it relies on the phase shift of the X-ray wave front caused by
79 inhomogeneities in the sample, as opposed to absorption in conventional X-ray imaging.
80 Edge Illumination (EI) XPCi is a differential phase imaging technique that uses a coded
81 aperture system to translate the change in the X-ray propagation direction that
82 accompanies the phase effects into a variation in the detected intensity [17]. With the
83 acquisition of at least three projection images, this method allows for the simultaneous
84 retrieval of standard absorption, refraction, in which the pixel intensity represents the
85 refraction angle of the beam, and scattering images: the latter are a complementary
86 representation of the microscopic structure of the imaged sample, which highlights
87 ultra-small angle scattering caused by features in the sub-pixel scale [18]. Several other
88 XPCi imaging techniques exist, e.g., Talbot-Lau grating interferometry was used to
89 investigate impact damage in composite plates [19]. However, EI offers several
90 advantages in terms of practical implementation in fast, large field of view, and
91 vibration-resistant industrial systems [20].

92 XPCi of composite plates performed using synchrotron radiation was observed to be
93 sensitive to variations in homogeneity [21], and enabled the detection of cracks in
94 macro-fibres and voids to much higher standards than conventional radiography [22].
95 Talbot-Lau grating interferometry XPCi was used to characterise the weave pattern, size
96 of fibre bundles, and resin contents of pre-preg CFRP samples [19]. The Talbot-Lau
97 approach was also used for the study of fibre orientation in short glass fibre reinforced
98 composites using the direction-dependent properties of ultra-small-angle scattering
99 [23].

100 Limited research has compared ultrasonic and X-ray phase contrast imaging
101 techniques for damage detection in fibre-reinforced composite plates. Gresil et al. [24]
102 used immersion through-transmission imaging and phased array pulse echo imaging to
103 benchmark Talbot-Lau grating interferometry XPCi as a viable NDE technique.
104 Measuring porosity in pre-preg CFRP samples, they concluded that XPCi offered a much
105 better quantification than ultrasonic phased array or through-transmission techniques.
106 Characterisation of low velocity impact damage in a cross-ply composite plate was
107 performed using both ultrasonic C-scan imaging and EI XPCi. It was shown that the
108 phase-enhanced and scattering images presented an advantage for the detection of
109 small defects (tens of μm and below), as opposed to the ultrasonic C-scan imaging,
110 which was capable of detecting macro-defects such as delaminations (100 μm and
111 above) [25].

112 In this contribution, EI XPCi and ultrasonic C-scan imaging were used to assess the
113 damage in a small pre-preg CFRP specimen with impact damage, and compare the
114 features observed using the two imaging methods. EI XPCi CT scans and ultrasonic C-

115 scans of the sample were taken, and the images were qualitatively analysed, to identify
116 and quantify the extent and nature of the observed damage, as well as any additional
117 feature revealed by EI XPCi. Confirmation of the observed features was achieved using
118 a commercial high-resolution X-ray CT scan.

119 **2. Experiments**

120 A 2 mm thick, 19 mm x 19 mm carbon fibre/epoxy resin cross-ply laminate sample
121 containing severe impact damage was imaged. The 16 plies were measured to be
122 approximately 150 μm thick, and the top and bottom laminae had a woven structure.
123 The sample contained an indent induced by impact, approximately 4 mm in diameter
124 and 1.2 mm in depth, which resulted in a small protrusion on the back of the plate,
125 approximately 5.5 mm in diameter and 0.5 mm thick, as shown in Fig. 1.

126 Ultrasonic immersion C-scans were performed using a $\frac{1}{4}$ inch (6mm) diameter,
127 focused longitudinal 20MHz transducer. The transducer was mounted on a computer-
128 controlled scanning rig, perpendicular to the surface of the sample (Fig. 2). The
129 transducer was positioned at the focal length of $\frac{3}{4}$ inch (19mm) above the surface of the
130 sample. A pulser/receiver (Panametrics 5601T) was used to drive the transducer and
131 amplify the pulse-echo signal. For each scan point the full A-scan was acquired and
132 saved to MATLAB using a digital storage oscilloscope (LeCroy 9304) with a sampling
133 frequency of 100 MHz. A 10mm thick steel plate was placed at the bottom of the water
134 tank and the composite sample was placed 3 mm above the surface of the steel plate,
135 as shown in Fig. 2, to allow for double-through transmission signals to be recorded. The
136 scans were performed over an area of 130 by 130 steps, with a 200 μm step size in both
137 the width and length directions, to contain the full sample. The scanning time was about

138 8 hours on a laboratory system. The transducer focal spot was calculated to be 280 μm ,
139 the wave velocity was estimated as 1380 m/s, corresponding to 70 μm wavelength. The
140 A-scans at different lines along the sample were used to generate B-scan visualizations
141 for comparison with EI XPCi CT slices.

142 The experimental setup used to acquire EI XPCi CT images of the sample included
143 a Rigaku MicroMax 007 HF rotating anode molybdenum X-ray source, with a 70 μm focal
144 spot, used at 40 kVp and a current of 20 mA. The detector used was a Hamamatsu
145 C9732DK flat panel CMOS detector with a 50 x 50 μm^2 pixel size. Due to cross-talk
146 between neighbouring detector pixels caused by the diffusion in the scintillator, the
147 effective resolution of the detector was approximately 100 μm [26]. EI XPCi was
148 implemented using a set of coded aperture systems (masks): a first mask, referred to as
149 the sample mask, was placed upstream of the imaged sample, splitting the divergent
150 beam into an array of individual beamlets (Fig. 3). The sample mask had an aperture of
151 12 μm and a period of 38 μm , with the source to sample mask distance (Z_{SM}) of 0.65 m
152 and 0.05 m upstream of the sample stage (Z_{OM}). A second mask, referred to as the
153 detector mask, was placed in contact with the detector, making the regions separating
154 adjacent pixels insensitive to incoming X-rays. The detector mask had a 20 μm aperture
155 and a 48 μm period, and the source to detector distance (Z_{SD}) was 0.85 m. The system
156 magnification was calculated to be 1.25. The masks were made of gold on a graphite
157 substrate and were manufactured by MicroWorks. The resulting system was only
158 sensitive to the phase effects in the x-direction, where a gradient in the refractive index
159 of the sample results in the refraction of the beamlets and thus leads to a change in the
160 detected intensity [17].

161 The CT image acquisition process included the acquisition of the illumination
162 curve (typically assumed to be a Gaussian), which represents the variation of the
163 detected intensity as a function of the sample mask position relative to the detector
164 mask position [17]. Five relative mask positions, translating the sample mask along the
165 x-direction in sub-pixel steps of 6 μm , were used during the CT acquisition to minimise
166 the scan time. 5 sets of 1800 projections, with a rotation of 0.2° per projection, were
167 taken at each mask position. Each projection had 1.2s exposure time, resulting in an
168 overall acquisition time of 12 hours (3 hours live scanning time) due to overheads in the
169 non-optimised system. The voxel size was $(41 \mu\text{m})^3$. Phase retrieval was carried out on
170 the image sequence to obtain the absorption, refraction, and scattering projection
171 images, which were then reconstructed using a CT reconstruction algorithm provided by
172 Nikon.

173 The standard phase retrieval method, discussed in [18], fits a Gaussian
174 distribution to the measured intensity of each pixel in the images with the sample. The
175 Gaussians were then compared to the correspondent Gaussians fitted without the
176 sample, as shown in Fig. 4. The absorption images were retrieved from the decrease in
177 amplitude (A_0 to A_1), the refraction images from the change in the centre position of the
178 Gaussian, and the scattering images from the change in the width of the Gaussian,
179 caused by ultra-small-angle scattering [18]. To take the cross-talk between neighbouring
180 apertures into account, a new phase retrieval method was used [27], which involves
181 fitting three overlapping Gaussian distributions, and using the coefficients of the central
182 Gaussian for the phase retrieval. The reconstructed CT images were then segmented
183 and visualised using Drishti, a volume exploration and presentation tool [28].

184 An additional high resolution X-ray scan of the sample was performed using a
185 commercial X-ray system (Nikon XTEK XTH 225kV). The scan was performed using 40kV
186 beam energy, 358 μ A beam current and a power of 14.3W. A PerkinElmer 1620 flat panel
187 detector was used, with 200 μ m pixel size. The system had a geometric magnification of
188 14 resulting in an effective pixel size of 14 μ m. The exposure time per projection was 1s,
189 and a total of 3185 projections were acquired, resulting in a scanning time of
190 approximately 1 hour. The experimental setup for this system is described in [29].

191

192 **3. Results and Discussion**

193 The results obtained from the ultrasonic C-scans and EI XPCi CT scans are presented
194 as slices taken from different cross-sections of the sample. For each cross-section, two
195 B-scans are presented with, respectively, the front and back surface of the sample facing
196 the transducer, with the front surface containing the indent, and the back surface the
197 protrusion. Three retrievals (absorption, refraction and scattering) are shown for the
198 matching CT slice, as well as an RGB image, which is the superposition of the three
199 retrieved images, with absorption in blue, refraction in green and scattering in red.
200 Three sets of figures are shown, one from an undamaged area of the sample, one
201 showing a delamination across the sample, and one from the most severely damaged
202 part of the sample.

203 **3.1 Undamaged Part**

204 For an undamaged part of the composite sample, the ultrasonic B-scans (Fig.
205 5a/b) and the X-ray images (Fig. 5d/f) show reasonably uniform ply layers. The ultrasonic

206 B-scan images show strong reflection from the surface and back wall, with weaker
207 reflections from the inner plies. Due to surface unevenness, the reflection from the front
208 (impact side, Fig. 5a) is larger and more irregular than that from the back surface (Fig.
209 5b). The reflections from the inner plies in both B-scans indicate reasonably
210 homogeneous and aligned plies, but potentially with some waviness.

211 The matching X-ray absorption CT slice (Fig. 5d) shows contrast due to the cross-
212 ply layup of the sample, with regular plies visible. The contrast can be enhanced (Fig. 5g)
213 for the uneven sample surfaces to be observed better. Plies are clearly defined, with a
214 signal from the ply interfaces visible in the refraction image (Fig. 5e), with the intensity
215 of the intra-ply area matching the background grayscale. This points toward a
216 homogeneous distribution of plies and ply alignment. This homogeneity is more clearly
217 visible in the 3D rendering of the refraction signal in Fig. 5h, where the contrast was
218 adjusted to highlight the interfaces in the sample, resulting in a strong signal from the
219 surfaces of the sample and a lack of signal from the intraply area within the sample. The
220 only visible signal is due to the impact damage at the center of the sample, as opposed
221 to the inhomogeneities from the ply structure.

222 The scattering images (Fig. 5f/i) show clear signals from the ply layer interfaces,
223 suggesting small, sub-pixel inhomogeneity. Figure 5e/f shows some imaging artefacts in
224 the CT reconstruction at the specimen edges due to the specimen shape. The 3 X-ray
225 retrieved images were superimposed in Fig. 5j to highlight the complementarity of the
226 signals. For the undamaged part of the sample, the edges are clearly visible, and an
227 indication of the ply layers can be observed from the absorption (blue) and scattering
228 signal (red).

229

230 **3.2 Delamination**

231 Figure 6 shows a cross-section about 5 mm from the impact center, where no
232 indent can be observed. Delamination is detected by the ultrasonic B-scan and X-ray
233 images close to the bottom of the sample. The delamination was localized between plies
234 14 and 15 from the X-ray images, in agreement with the ultrasonic B-scan images, as the
235 first reflection from the delamination in both scans was found to be approximately 0.3
236 mm from the back surface of the sample, which corresponds to the thickness of the two
237 plies. In the B-scans, a strong reflection was observed (Fig. 6a/b), characteristic of a
238 delamination [9]. From the X-ray retrievals, the delamination length at that location was
239 measured to be approximately 5 mm in length from the absorption and refraction
240 images. This was in good agreement with the measurement of approximately 5.5 mm
241 from the ultrasonic B-scan, considering the system resolution (0.2mm step size).

242 In the slices and the 3D rendering of the X-ray images, the delamination is visible
243 in all three retrievals, with the absorption images showing a separation between the
244 plies (Fig. 6d, with stretched contrast in Fig. 6g). A strong signal is observed in the
245 refraction retrieval (Fig. 6e/h) due to the interface created between the delamination
246 and the neighbouring plies. A strong scattering signal is also observed around the
247 delamination (Fig. 6f/i), indicating the presence of the sub-pixel damage. Identification
248 of the features observed in the scattering signal is not directly achievable, but it could
249 indicate that either the delamination (ply-separation) extends further than observed in
250 the absorption and refraction signal (Fig. 6j), or additional micro-damage e.g.
251 debonding, micro-matrix cracks, or fibre damage. These features are better visible in the

252 3D renderings shown in Fig. 6h/i with enhanced contrast to highlight the interfaces of
253 the delamination in the refraction signal and the surrounding damage in the scattering
254 image, respectively. These micro-features are unique to the scattering signal and offer
255 a more accurate representation of the possible damage extent, which could not be
256 observed in the other signals due to the scale of those features. The complementarity
257 of the scattering signal allows to observe features of different magnitudes and nature
258 and thus to better understand the defects present in the sample. The 2-dimensional cuts
259 through the 3D rendering of the X-ray signals (Fig. 7) shows that the delamination is not
260 complete, but is shaped as a ring around the impact damage which pushed the central
261 part of the specimen downwards and caused the protrusion. The refraction and
262 scattering renderings show in addition to the main delamination towards the back of
263 the specimen, smaller interfaces and delaminations within the protrusion.

264

265 **3.3 Center of Damage**

266 The impact caused damage across the sample, as can be seen from both the
267 ultrasonic B-scans (Fig. 8a/b), and the X-ray images across the impact center. The
268 ultrasonic B-scans exhibit one of the limitations of ultrasonic testing for damage
269 detection in composite plates. The indented front surface and protrusion cause strong
270 reflections and scattering, with potentially unobserved internal defects due to lack of
271 signal past the damage closest to the surface. An example of such a situation is observed
272 here, where a crack across the thickness of the sample was only observed in the X-ray
273 images (Fig. 8). The macro-crack was observed in all three X-ray retrievals, however they
274 indicate different features; the absorption CT slice (Fig. 8d) shows clear damage in the

275 sample, with the crack and further damage better visible with enhanced contrast (Fig.
276 8g). The refraction signal (Fig. 8e/h) highlights the interface of the crack through the
277 sample, whereas the scattering signal (Fig. 8f/i) indicates further micro-damage in the
278 sample surrounding the crack. This level of accuracy in the measurement of the damage
279 extent is unachievable using ultrasonic imaging or conventional attenuation-based CT
280 alone, represented here by the retrieved absorption images (Fig. 8d).

281 Moreover, a clear scattering signal can be observed throughout the damaged
282 area of the sample. The presence of signal in the different retrievals indicates the
283 presence of different types of damage (Fig. 8j). The signal in the refraction images is due
284 to the small voids created by the separation of plies in the inter-ply area originating from
285 the material being displaced when the damage occurred, as shown in Fig. 8k. The
286 scattering signal is due to the micro-damage that occurs within the plies due to the
287 material displacement. The multimodal imaging of the sample, as well as the
288 superposition of all three retrievals in an RGB image, as shown in Fig. 8j/k, thus allows
289 to locate and clearly visualise the damage, as well as identify the scale of damage
290 involved and have a more accurate representation of the extent of the damage from the
291 refraction and scattering images to complement the absorption and ultrasonic images.

292

293 **3.4 Quantification of damage extent and comparison with high-resolution X-** 294 **ray absorption CT scan**

295 A quantification of the damage extent was achieved using a 2D projection of the
296 CT slices through the specimen thickness for all three X-ray signals, effectively adding
297 the damage indications to visualize the overall damage extent. It can be observed that

298 in the absorption 2D projection (Fig. 9a) the damage appears round-shaped, similar to
299 the shape observed in the double through transmission ultrasonic C-scan (Fig. 9g).
300 However, the refraction and scattering signal show a more square-shaped damage
301 indication (Fig. 9c/e), due to the types of defects observable in these signals. Profile
302 plots were taken across the 2D projections of the sample in order to assess the damage
303 extent across the damaged area of the sample. Examples of such profile plots are shown
304 in Fig. 9b/d/f/h, corresponding to the red lines in the 2D projections. A simple
305 thresholding method was used to assess the extent of the damage. It was measured to
306 be approximately $5.9 \times 5.4 \text{ mm}^2$, $5.9 \times 5.7 \text{ mm}^2$ and $6.6 \times 7.3 \text{ mm}^2$ for the absorption,
307 refraction and scattering signals, respectively. As expected, the scattering signal
308 indicates a larger damage extent than the absorption and refraction signals, due to its
309 sensitivity to sub-pixel features. The dimensions observed in the double-through
310 transmission C-scan of the sample were calculated to be approximately $6.7 \times 7.1 \text{ mm}^2$,
311 comparable to the size obtained from the scattering signal.

312 The features observed in the EI XPCi signals were compared to a high resolution
313 X-ray CT scan performed using a commercial system with voxel size of $(14\mu\text{m})^3$. The
314 higher resolution absorption scan (Fig. 10b) confirms the features observed from the
315 lower resolution absorption (Fig. 10a), refraction, and scattering signals. This is more
316 clearly visible in Fig. 10c/d, where the respectively the refraction and scattering signals
317 were superimposed on the high resolution absorption scan. The interfaces observed in
318 the refraction signal are confirmed by the high resolution scan (Fig. 10c arrows). The
319 presence and extent of those interfaces confirms our interpretation as to the nature of
320 the refraction signal. Small voids created between the plies, not visible in the low
321 resolution absorption scan (Fig. 10a), became visible in the high resolution absorption

322 scan (Fig. 10b), and match the superimposed scattering signal (Fig. 10d). The scattering
323 signal extends beyond the damage visible in the high resolution scan in certain areas,
324 suggesting that an even higher resolution scan would be needed to observe these
325 features. This confirms our hypothesis that the scattering signal indicates micro-damage
326 at a scale below the resolution of the imaging system.

327

328 **4. Conclusions**

329 A qualitative comparison between ultrasonic immersion C-scan imaging and EI
330 XPCi images was performed on a small, cross-ply composite sample with severe impact
331 damage to investigate the different features observable using the two imaging
332 techniques. Standard ultrasonic C-scan imaging allowed the detection and sizing of the
333 overall damage. The delamination close to the bottom layer was accurately sized with
334 good contrast from the C-scans. However, two main limitations of ultrasonic scanning
335 were observed. Good penetration depth and a clear reflection of the respective back
336 wall was seen, but the chosen ultrasonic frequency (20 MHz) corresponded to a
337 wavelength comparable to the ply layer thickness. Together with a non-smooth sample
338 surface, this made accurate measurement of ply layer thickness and waviness difficult.
339 The second limitation can be seen in Fig. 6a/b and Fig. 8a/b, where strong reflections at
340 large defects (e.g. delamination) prohibited the detection of additional, internal defects.

341 EI XPCi resolves some of these limitations by offering a higher resolution as well
342 as visualisation of the full sample. As a result, EI XPCi allows the detection of internal
343 defects and features that are not visible using ultrasonic imaging, more specifically in
344 this case, a crack through the thickness of the sample. The multimodal imaging using EI

345 XPCi, resulting in the retrieval of absorption, refraction and scattering images of the
346 sample, contributes to the identification of the defects occurring in the damaged area.
347 The refraction signal highlights interfaces due to the separation of plies (small voids and
348 delaminations), and the scattering signal corresponds to sub-pixel features which
349 indicate micro-damage accompanying the main defects. Combining the features
350 observed from the different X-ray retrievals, an accurate estimation of the damage
351 extent can be obtained from a low resolution system relative to ultrasonic imaging or
352 higher resolution conventional radiography. Based on the physics of the different
353 signals, some assessment of the detected damage type can be obtained, but below the
354 system resolution only the presence of damage can be obtained from the scattering
355 signal. However, EI XPCi CT imaging currently has some limitations, as the scanning
356 times are long and, similar to conventional CT imaging, the sample must be small enough
357 to be contained in the field-of-view. This could be resolved by scanner acquisition
358 optimisation and by the introduction of computed laminography for further
359 investigations.

360

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362

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368

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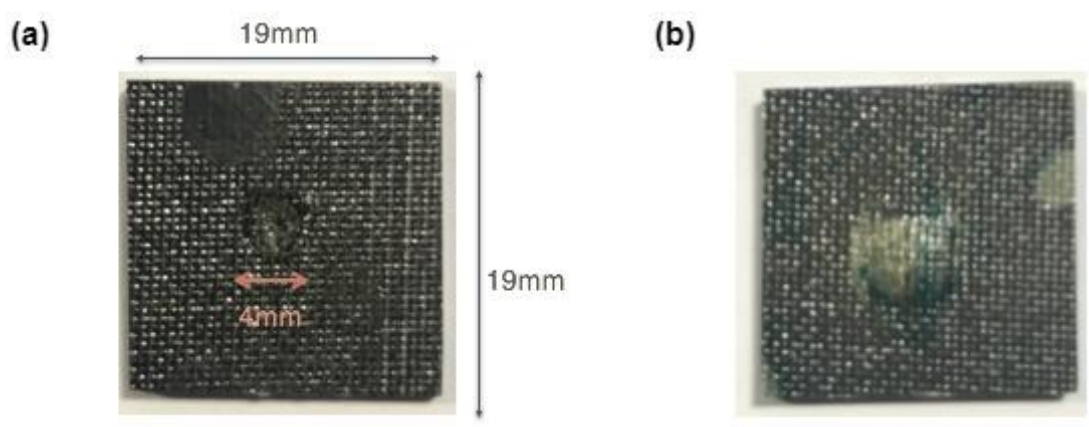
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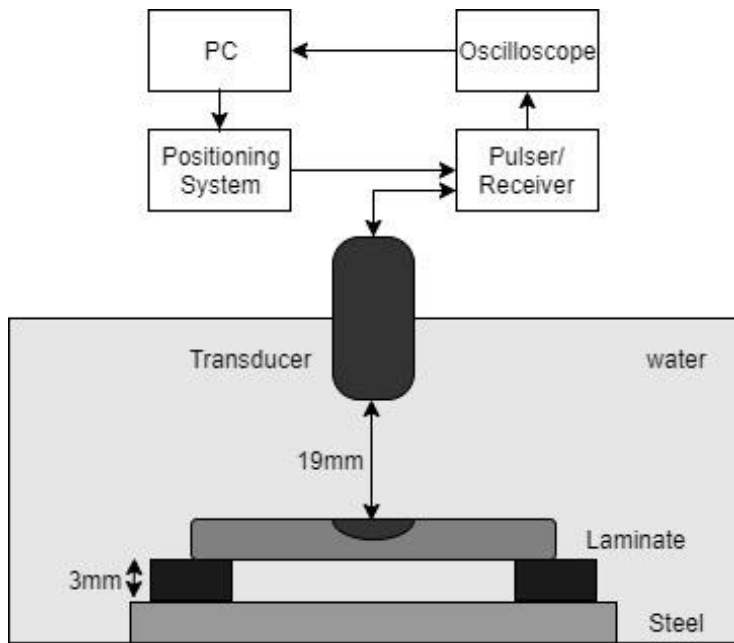
500 **Figure 1**



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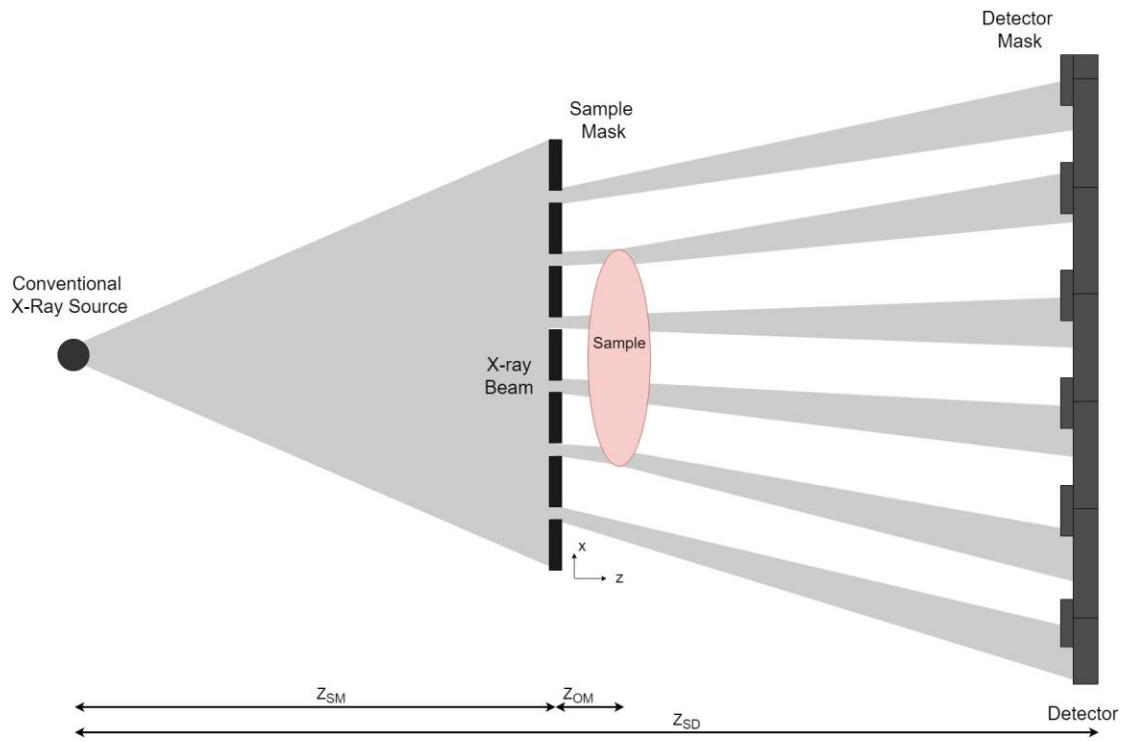
503 **Figure 2**



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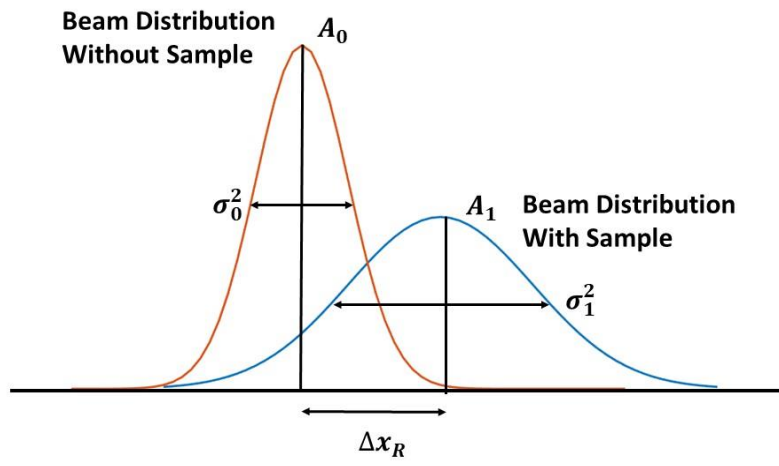
Figure 3



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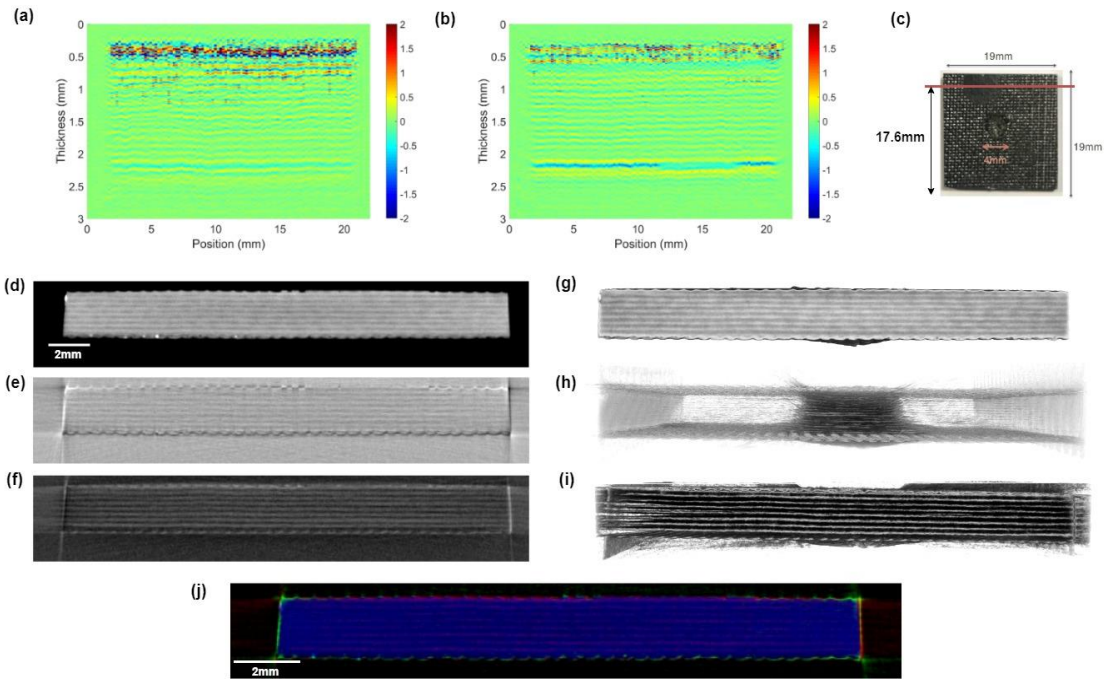
Figure 4



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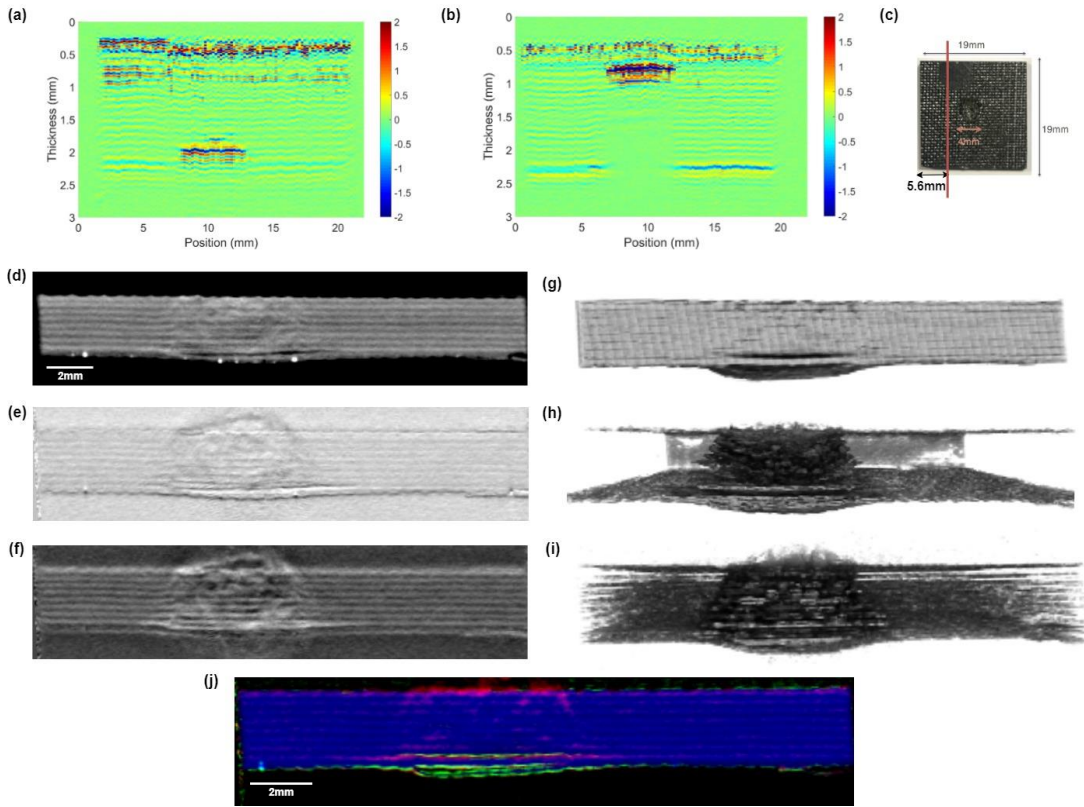
510 **Figure 5**



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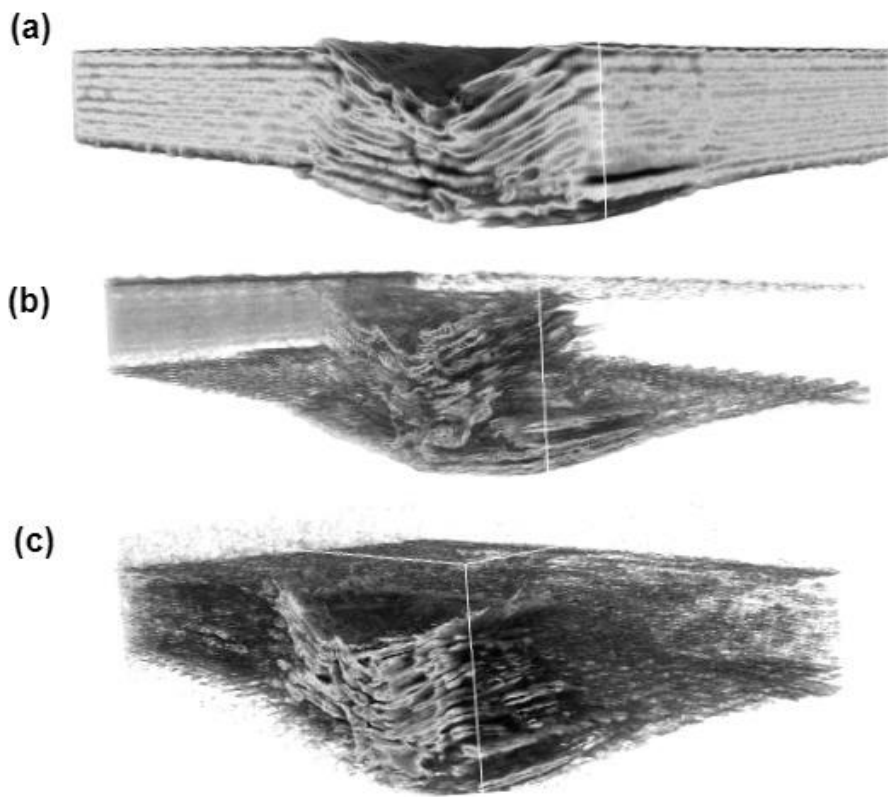
Figure 6



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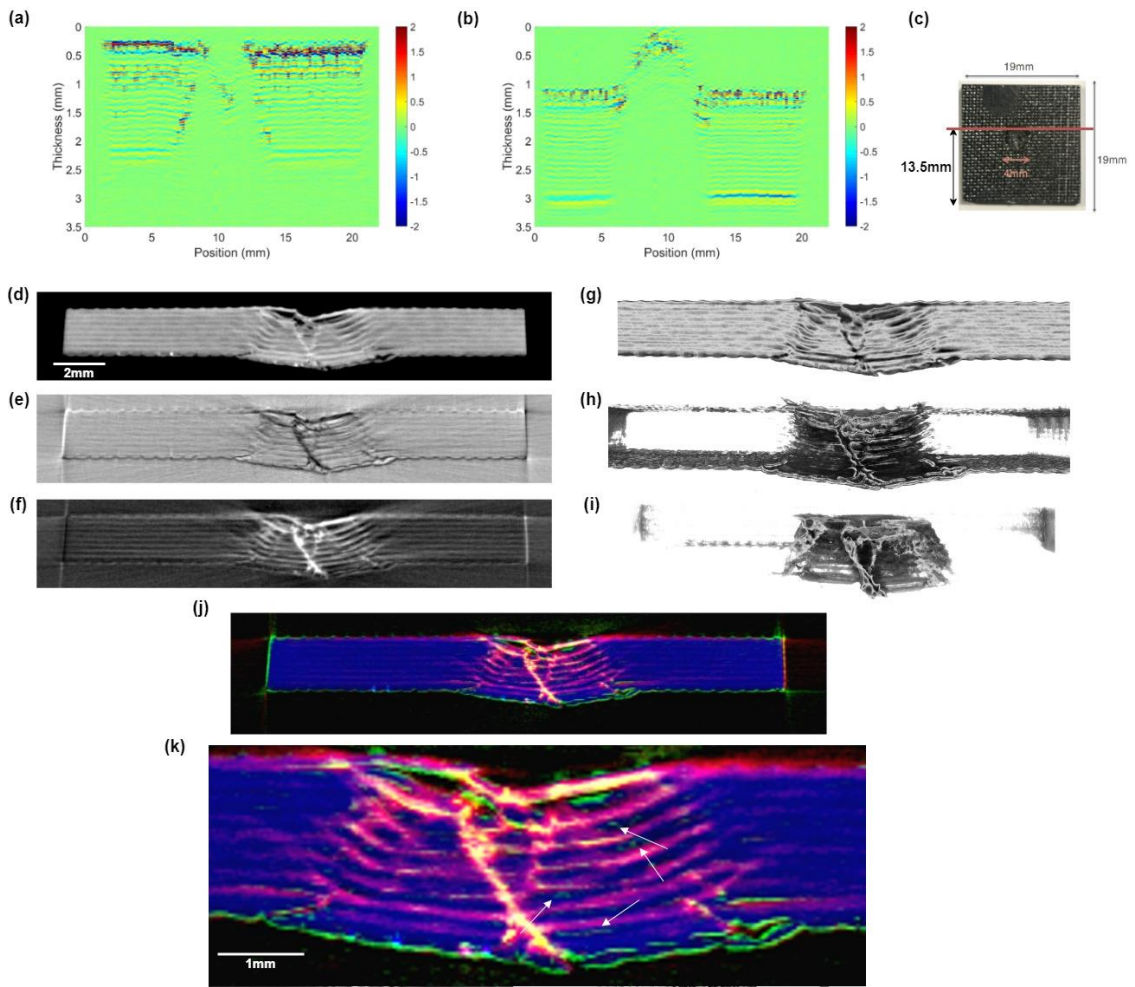
515 **Figure 7**



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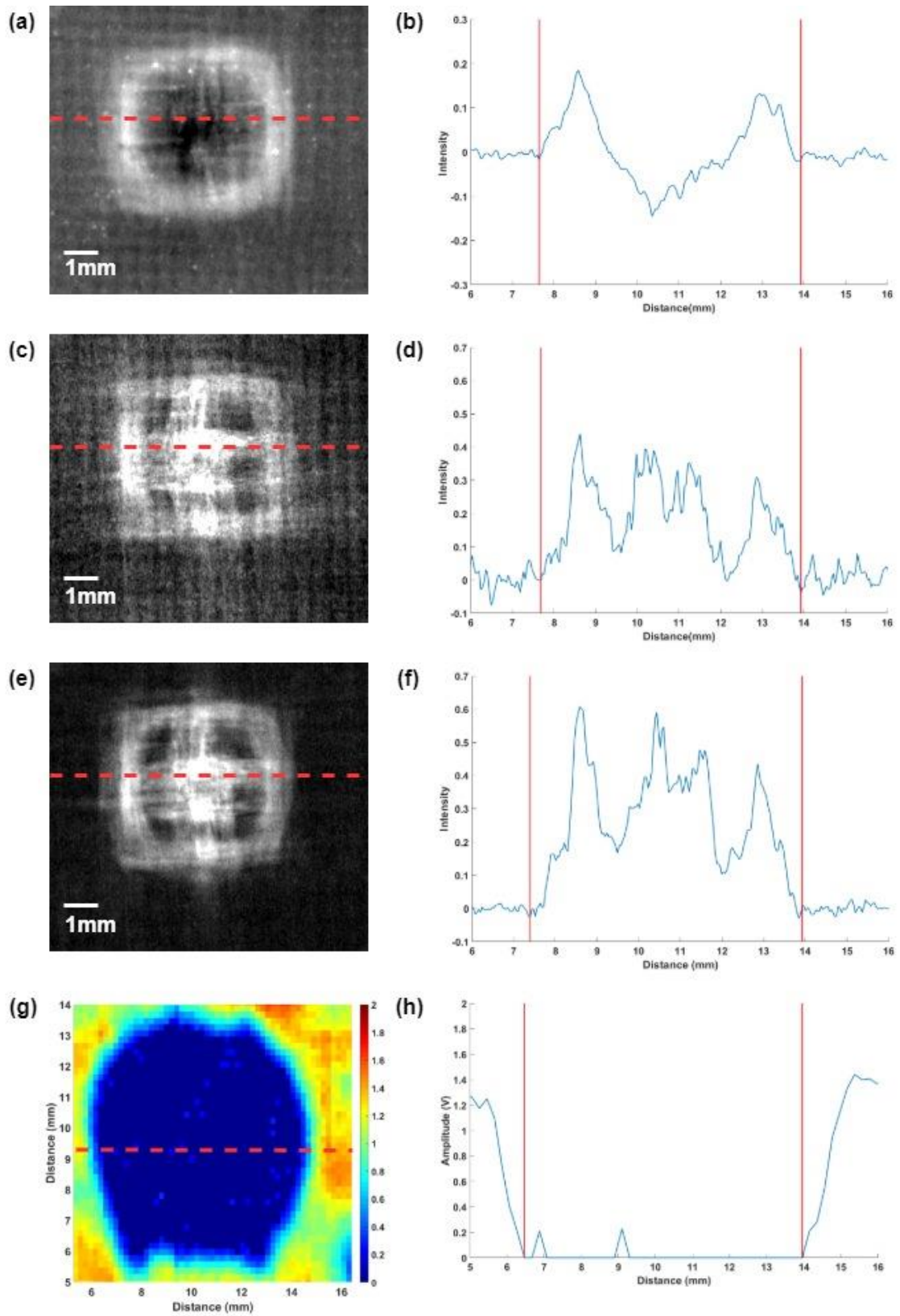
518 **Figure 8**



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521 **Figure 9**



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