1 2	Numerical modelling of interactions of waves and sheared currents with a surface piercing vertical cylinder	
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18	Highlights	
19	• Two different approaches, 2-D OpenFOAM and Lagrangian wave-current simulations, are us	sed
20	to model focussed wave groups and sheared currents simultaneously in a controlled mann	
21 22	and produce input conditions for 3-D OpenFOAM models to investigate wave-current-structu interactions.	ure
23	<ul> <li>Good agreement between numerical results and experimental data is obtained, indicating the</li> </ul>	hat
24	both approaches are capable of replicating experimental wave-current flows, and accurate	
25	modelling interactions between surface piercing cylinders and focussing waves on shear	red
26	currents.	
27 28	<ul> <li>The performance of both approaches is evaluated in terms of accuracy and computational effort required.</li> </ul>	ort
29	• It is found that the method of coupling the 3-D CFD and Lagrangian models is computation	nal
30	slightly cheaper and more accurate because of the use of a smaller computational domain a	
31	the iterative wave-current generation in the faster Lagrangian model.	
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33 Abstract

34 Vertical surface piercing cylinders, such as typical coastal wind turbine foundations and basic elements 35 of many coastal structures, are often exposed to combined loading from waves and currents. Accurate 36 prediction of hydrodynamic loads on a vertical cylinder in a combined wave-current flow is a 37 challenging task. This work describes and compares two different approaches for numerical modelling 38 of the interaction between focussed wave groups and a sheared current, and then their interactions with 39 a vertical piercing cylinder. Both approaches employ an empirical methodology to generate a wave 40 focussed at the location of the structure in the presence of sheared currents and use OpenFOAM, an open source Computational Fluid Dynamics (CFD) package. In the first approach, the empirical wave-41 42 on-current focussing methodology is applied directly in the OpenFOAM domain, replicating the 43 physical wave-current flume. This approach is referred to as the Direct Method. In the second approach, 44 a novel Lagrangian model is used to calculate the free surface elevation and flow kinematics, which are 45 then used as boundary conditions for a smaller 3-D OpenFOAM domain with shorter simulation time. This approach is referred to as the Coupling Method. The capabilities of the two numerical methods 46 have been validated by comparing with the experimental measurements collected in a wave-current 47 48 flume at UCL. The performance of both approaches is evaluated in terms of accuracy and computational 49 effort required. It is shown that both approaches provide satisfactory predictions in terms of local free

50 surface elevation and nonlinear wave loading on the vertical cylinders with an acceptable level of 51 computational cost. The Coupling Method is more efficient because of the use of a smaller 52 computational domain and the application of the iterative wave-current generation in the faster 53 Lagrangian model. Additionally, it is shown that a Stokes-type perturbation expansion can be 54 generalized to approximate cylinder loads arising from wave groups on following and adverse sheared 55 currents, allowing estimation of the higher-order harmonic shapes and time histories from knowledge 56 of the linear components alone.

- 57
- 58 Keywords

59 Focussed wave groups; sheared currents; wave-on-current focussing methodology; Lagrangian wave-60 current flume; Harmonic reconstruction; OpenFOAM;

- 61
- 62 1. Introduction

The review articles by Peregrine and Jonsson (1983a; b), Thomas and Klopman (1997) and Wolf and Prandle (1999) have shown that wave-current interaction is one of the important physical processes in coastal waters. The presence of a background current modifies the wave dispersion, wave-induced velocities and shear stress near the seabed etc., so has an effect on wave loads on structures and wave propagation near coastlines. Coastal engineering applications, such as the design of coastal protection and structures as well as the evaluation of sediment transport and coastal erosion, would benefit from an enhanced knowledge of this complex process and its effect on coastal structures.

70 In existing design methods, the current profile is usually assumed to be uniform with depth. The uniform 71 current approximation may apply for large-scale ocean currents and deep tidal flows, but it fails to 72 model wind-driven currents and tidal flows in shallow coastal waters that exhibit some degree of 73 variation in the vertical direction (Chakrabarti, 1996; Forristall and Cooper, 1997; Stacey et al., 1999; 74 Gunn and Stock-Williams, 2013). Previous studies demonstrated that the velocity shear modifies the 75 wave dispersion relation (Swan et al., 2001a), produces changes in water-surface elevation (Tsao, 1959; 76 Brink-Kjaer, 1976; Kishida and Sobey, 1988), and causes significant effect on the tendency of surface waves to break (Peregrine and Jonsson, 1983; Yao and Hu, 2005) in a different way when compared to 77 78 depth-uniform currents. This work considers a current profile which varies with depth so has a 79 significant depth-varying vorticity distribution. Such a profile is a more realistic representation of a 80 current flow in some regions in the open sea.

81 The vorticity dynamics due to wave-shear current interaction can be described by the vorticity transport 82 equations, which are obtained by taking the curl of the momentum equations. Analytical solutions of 83 the vorticity transport equations exist only for the constant-vorticity case (the current is linearly sheared) 84 (Thomas, 1981; 1990; Nwogu, 2009). For more realistic profiles that vary arbitrarily with depth, the 85 computation is more difficult because of the changing vorticity field in space and time. For initially uniform vorticity, Kelvin's circulation theory applies and the vorticity remains uniformly distributed. 86 87 Then the wave motion can be treated as an irrotational disturbance, as described by Teles Da Silva and 88 Peregrine (1988). Approximations are necessary if analytical solutions are to be sought for the cases 89 with arbitrary vorticity. Various techniques have been developed (Kirby and Chen, 1989; Swan and 90 James, 2001; Ko and Krauss, 2008; Smeltzer and Ellingsen, 2017), yet these have limited range of 91 applicability; the wave is linear or weakly nonlinear, and the current strength lies within a certain range 92 (either weak, moderate or strong). The difficulties inherent to problems associated with strongly sheared 93 currents have necessitated the use of Computational Fluid Dynamics (CFD), which is a promising tool 94 for modelling the interactions between waves and current, and both with structures.

95 Much previous numerical work based on CFD has primarily concentrated on regular wave interactions with currents (Santo et al., 2017; Zhang et al., 2014; Markus et al., 2013; Li et al., 2007; Park et al., 96 97 2001). However, Tromans et al. (1991) suggested the use of NewWave-type focussed wave groups as 98 design waves representing individual extreme events in random seas. Jonathan and Taylor (1997), 99 Taylor and Williams (2004), Santo et al. (2013), Christou and Ewans (2014), among others, confirmed 100 that this theory is applicable to a wide range of wave conditions. The original NewWave theory was developed for deep water waves. Later it was demonstrated that it can be applied to waves on shallow 101 102 water (Whittaker et al., 2016). The use of NewWave-type wave groups for wave-structure interaction 103 has been demonstrated by Zang et al. (2006, 2010) for a ship-shaped fixed body and for a surface piercing cylinder, respectively. Further work using wave groups on cylinders is described in the papers 104 105 by Fitzgerald et al. (2014) and Chen and co-workers (2014, 2016, 2018), and for jacket-type structures 106 in Santo et al. (2018).

Wind turbines with cylindrical foundations are likely to be located in areas with severe wave conditions, with intermediate and shallow water depths and with significant currents generated by tides, storm wind shear etc. Thus, the interaction of focussed wave groups propagating on either following or adverse sheared currents with surface-piercing cylinders has direct practical applications.

111 The primary challenge in the numerical modelling of focussed wave groups on sheared currents is the

simultaneous and controlled generation of focussed wave groups on flow with non-uniform vorticity.

113 The co-existence of waves and currents alters both the evolution of the waves and the profile of the

114 currents in a way unpredictable by existing analytical approaches. As such, neither the point of focus

nor the elevation of the wave and the underlying flow field are known a-priori.

116 Various approaches are used to achieve wave focussing at a particular location and time in the absence 117 of currents, including a dispersive focussing method and various iterative techniques. The dispersive 118 focussing method calculates the initial phase shift of each wave component based on linear wave theory. 119 This inevitably results in a shift of the actual focus position due to non-linear wave-wave interactions (Rapp and Melville, 1990; Baldock et al., 1996; Johannessen and Swan, 2001). The iterative methods 120 121 reconcile this issue by iteratively correcting either only the initial phases (Chaplin, 1996; Yao and Wu, 122 2005) or both the initial phases and the amplitudes (Schmittner et al., 2009; Fernandez et al., 2014; 123 Buldakov et al., 2017) of different wave frequency components in a wave group. The iterative approach 124 derived in Buldakov et al. (2017) calculates the corrected input for the wavemaker considering only the 125 linearized part of wave spectrum and therefore it differs from any previous methodology. This approach 126 has been successfully applied to physical experiments of focussed wave groups on sheared currents 127 (Stagonas et al., 2018a). The wave focussing methodologies discussed previously were mainly used in physical experiments; however, its application to a numerical wave flume is straightforward and can be 128

implemented in a similar way to that in a physical flume (Stagonas et al., 2018b).

130 For either 2-D or 3-D CFD simulations, a computationally expensive fine grid is necessary to accurately 131 resolve the non-linear evolution of focussed wave groups, and the complex flow-structure interaction. Applying empirical wave-on-current focussing techniques in CFD-based models, even in 2-D, may 132 yield substantial increases of the computational effort required. To accommodate this, a faster numerical 133 134 model may be used alternatively to produce the input wave-current kinematics for CFD-based models. 135 This work describes and compares two CFD modelling approaches building on the widely used open-136 source CFD platform OpenFOAM. In the first approach, the wave-on-current focussing methodology (Stagonas et al., 2014; 2018a; 2018b) is applied directly to a CFD numerical wave flume, replicating 137 138 the physical wave-current flume. 2-D simulations are performed first to calculate iteratively the boundary conditions required to produce focussed wave groups on different flow conditions - namely, 139 140 quiescent flow without a current, adverse and following sheared current – and the interaction with the 141 structure is then modelled in 3-D. This approach is referred to as the Direct Method hereafter unless

142 otherwise stated.

143 In the second approach, a novel Lagrangian model (Buldakov et al., 2015) is coupled with the CFD 144 model. Differentiating it from recent one-way 'online' coupling approaches used to, e.g., model the 145 interaction of waves with cylinders (Paulsen et al. 2014a; 2014b), the time histories of the surface 146 elevation and flow kinematics are pre-computed using the Lagrangian model and are then used as inlet 147 boundary conditions for the CFD model. In this 'offline' coupling, all reflections are dealt by the CFD 148 simulation, eliminating the need for simultaneous computation and exchange of information between 149 the two models. This approach is referred to as the Coupling Method. We note that such a method of 150 domain decomposition, i.e. one-way coupling of simpler models with more advanced models, was also 151 applied by Biausser et al. (2004), Drevard et al. (2005), Christensen et al. (2009), among others for 152 various flow problems but excluding the effect of flow currents. Here, the 3-D numerical flume used in the Coupling Method is considerably shorter than that of the Direct Method, and the iterative wave-153 current generation is applied in the faster Lagrangian model. The performance of both approaches is 154 validated against experimental measurements and is evaluated in terms of accuracy and computational 155 156 effort. The rest of the paper is organized as follows. The physical experiments on wave-sheared currentcylinder interactions are described in Section 2. Details of the CFD and Lagrangian models are provided 157 in Section 3. The results of both numerical modelling methodologies are compared with the 158 159 experimental results in Section 4. Section 5 reconstructs the higher order harmonic forces using linear 160 components alone. Conclusions are given in Section 6.

161

### 162 2. Experimental setup and methodology

163 A set of experiments on wave-sheared current interactions with a vertical surface-piercing cylinder of

two different sizes was carried out and used to validate the proposed two CFD-based numerical models

165 in this work. This section describes the experimental setup and the applied methodology briefly.

166 2.1 Experimental setup

167 All experiments were conducted in a 20 m long, 1.2 m wide and 1 m deep recirculating wave-current 168 flume at University College London (UCL) with a water depth of 0.5 m. Two Edinburgh Design Limited 169 (EDL) force-feedback 'piston-type' wavemakers, one at each end of the facility, were used to generate 170 and actively absorb the waves. The flow entered vertically into the working section of the flume with 171 the inlet and outlet located approximately 1 m in front of each wavemaker, as shown in Figure 1. A 172 Cartesian coordinate system Oxz is introduced in both physical and numerical wave flumes such that 173 the origin O is the plane of the undisturbed free surface, x = 0 is the focus point, and z positive upwards.

The critical challenge of generating controlled and stable sheared currents was addressed through the use of two carefully designed flow conditioners/profilers installed on top of the inlet and the outlet. The conditioners/profilers consisted of 0.5 m long, 1.2 m wide and 0.88 m deep box sections consisting of vertically and horizontally placed cylindrical elements. Each cylindrical element had a diameter of 8 cm and was constructed using a 5 cm porous galvanised wire mesh, see Figure 2. Compared to previous work, the flow shaping approach used here has the comparative advantage of producing sheared currents with variable vorticity distribution without considerable interference to the generation of waves, see for

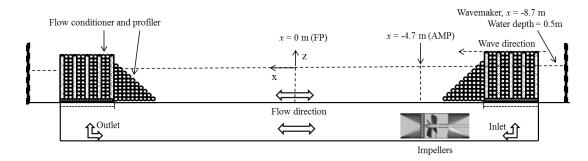
181 example Steer et al. (2017) and for more details see Stagonas et al. (2018a).

Flow kinematics were measured with a high speed, time resolved Particle Image Velocimetry (PIV) system produced by TSI Incorporated. The system employs a 5 W water cooled Argon Ion laser operated at a pulsating frequency of 1 kHz. A light arm was used to direct the laser sheet upwards through the bottom of the wave flume (the bed) and measurements were taken at the focus point (FP in Figure 1) and at a distance of approximately 27 cm from the side wall; these were also the locations of the free surface elevation measurements. The flow was seeded with 50  $\mu$ m polyamide particles and PIV images with a resolution of 1024×1024 pixels were recorded at a frame rate of 250 fps. An example of

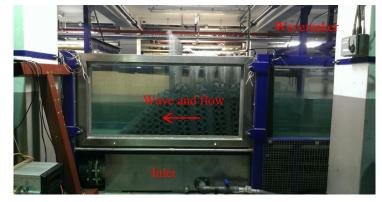
the kinematics measured for adverse and following currents will be given in the following section. The

190 velocity measurements are available from still water level (z = 0 m) to approximately 15 cm from the

- 191 bed, beyond which the camera view was blocked by the support structure of the flume. Surface elevation
- measurements are used not only for validation but also for providing the inlet boundary conditions for
- the numerical models.
- 194



- 196 Figure 1. Schematic side view of the UCL wave-current flume showing two wavemakers at each end
- 197 of the flume, and locations of inlet and outlet of the current discharge. FP stands for Focus Point, and
- 198 AMP means the location for amplitude matching.
- 199



200

- 201 Figure 2 Photograph of the conditioning and profiling system
- 202

203 Focussed wave groups were produced using a Gaussian target spectrum on a water depth of 0.5 m. The 204 same target spectrum was used for waves on adverse and following currents and without a current. The 205 peak frequency was set to 0.6 Hz and the point of focus was 8.7 m from the wavemaker (FP in Figure 206 1). The phases of different components in a wave group were forced to come to focus at the focus point 207 and the amplitudes were matched to the target spectrum at a distance of 4 m upstream of the focus point 208 (AMP in Figure 1). In this way, focussed wave groups with the same spectrum at a relatively short 209 distance (1 m) from the inlet were produced for all flow conditions. The evolution to focus was 210 measured in the physical wave-current flume using a set of wave gauges, providing the means to 211 validate the numerical results not only at the focus point but also in terms of the evolution of the wave 212 group along the flume.

- Free surface elevations in the flume were measured using 7 twin-wire resistance-type wave gauges positioned at x = -4.7 m, -3 m, -1.8 m, -1 m, -0.5 m, -0.25 m, and 0 m, and sampled at 100 Hz. A return period of 128 s and a focus time of 64 s were selected for the wave generation. Discrete input spectra consisting of 256 frequency components with  $\Delta f = 1/128$  Hz were used as input to the wavemaker. For
- simplicity, the wave groups produced were categorized based on the linear sum of the target amplitude
- 218 components,  $A_{\rm L}$ . Only the results of nonlinear wave groups with  $A_{\rm L} = 0.07$  m are used in the present
- work. The methodology employed to generate these wave groups and sheared currents both in the
- 220 physical and numerical wave flumes will be described in the following subsection.

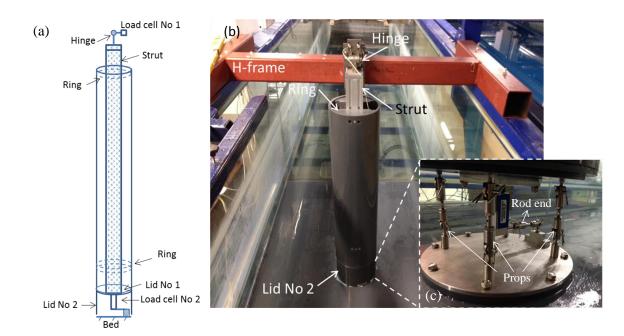




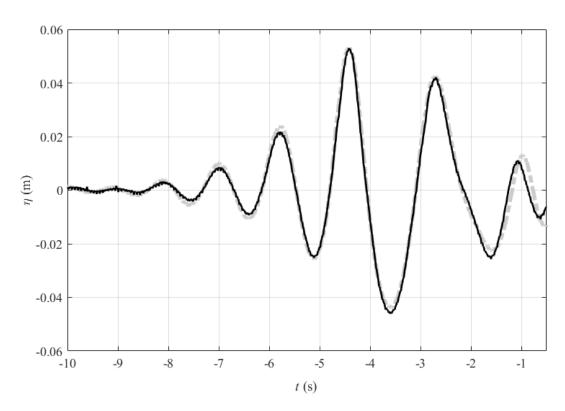
Figure 3 (a) Schematic diagram of side view of the cylinder model illustrating the location of the two load cells used to measure horizontal loads. (b) Photograph of the installed cylinder model. (c) Photograph showing the four props supporting the weight of the cylinder and the second/bottom load cell connected to the bed of the flume.

228 Experiments were also conducted with two different cylinders positioned at x = 0 m; for these cases an 229 additional wave gauge was placed at the front face of the cylinder. The smaller cylinder had a diameter 230 of 0.165 m and the larger one 0.25 m. For the smaller cylinder, flow induced loads were measured using 231 the load cell set-up described and used in Santo et al. (2017). However, in order to more effectively 232 support the weight of the larger cylinder, a different arrangement was developed to measure the fluid 233 induced horizontal force. The larger cylinder was a polyvinyl chloride (PVC) tube with a diameter of 234 D = 0.25 m. The PVC tube/cylinder was connected to an aluminium rectangular column, marked as 235 'strut' in Figure 3, via circular rings. The strut had dimensions of 0.09 m (breadth)  $\times$  0.09 m (width)  $\times$  1 236 m (height), and was connected to a load cell rated at 100 kg from the top through a hinge. This load cell, 237 labelled as Load cell No 1, was rigidly fixed on the steel H-frame that was in turn tightly fixed on the 238 flume walls. Another load cell, labelled as Load cell No 2, was located approximately 10 cm above the 239 flume's floor and was tightly fixed on the bottom of the cylinder/strut, see Figures 3(a) and (b). The 240 opposite end of the second load cell was connected through rod end bearings to an aluminium base, which was in turn fixed on the bed of the flume. Four props, also made using rod ends, supported the 241 242 weight of the structure resulting in a preload-free cell, see Figure 3(c). The overall arrangement 243 consisting of a strut, connecting rings and two load cells was mounted on rather than suspended from a 244 steel H-frame. The latter arrangement was used in this study for the smaller cylinder as aforementioned 245 and in previously reported tests by Santo et al. (2017).

A piece of PVC was used to model the bottom of the cylinder, labelled as Lid No 1, which was approximately 10 cm above the bed of the flume. Another piece of PVC, labelled as Lid No 2, was used to extend the model cylinder down to approximately 5 mm from the bed and compartmentalise the model cylinder, see Figures 3 (a) and (b). The compartment below the Lid No 1 was flooded and therefore a water-resistant load cell (Load cell No 2 in Figure 3) was used. Both load cells were sampled at 1 kHz and the experimental apparatus was calibrated for both tension and compression using dead weights at the beginning and the end of every testing cycle.

Surface elevation measurements recorded at x = -4.7 m (AMP in Figure 1) for different test cases illustrate a satisfactory level of repeatability. Representative results for experiments with waves on an 255 adverse current with the small (solid line) and the large (dashed line) cylinder in place are presented in 256 Figure 4. The repeatability in load cell measurements for the same testing conditions was also tested. 257 Standard deviations of 0.11 N/0.2 N were calculated from 15 horizontal force records acquired in 258 consecutive repeat tests with the smaller/larger cylinder exposed to waves on the adverse current. These 259 horizontal force results are representative of all the cases considered. It is to be noted that an iterative 260 methodology is used in both physical and numerical wave flumes to generate focussed wave groups 261 and sheared currents in a controlled manner. In the following sections, the iterative methodology is presented first and then the numerical flumes and implementation are described. 262





264

Figure 4 Example of free surface elevation time histories recorded at x = -4.7 m, for U = -0.2 m/s and  $A_L = 0.07$  m. Solid line: free surface elevation profile measured with the larger cylinder installed in the flume. Dashed line: free surface elevation profile measured with the smaller cylinder installed in the flume.

269

### 270 2.2 Generation of focussed waves on adverse and following currents

A methodology to accurately generate focussed waves without a current is described in Buldakov et al. (2017) and for waves on sheared currents in Stagonas et al. (2018b). The linearized part of the wave spectrum is isolated by linearly combining four non-linear free surface elevation time histories measured in the wave flume. Initially, a crest focussed wave is produced in the flume and the remaining three wave groups are generated with phase shifts of  $\pi$ ,  $\pi/2$  and  $3\pi/2$ . The measured spectrum (written as a complex variable a+ib) is then decomposed as

$$S_{0} = \frac{s_{0} + s_{1} + s_{2} + s_{3}}{4}$$

$$S_{1} = \frac{s_{0} - is_{1} - s_{2} + is_{3}}{4}$$

$$S_{2} = \frac{s_{0} - s_{1} + s_{2} - s_{3}}{4}$$

$$S_{3} = \frac{s_{0} + is_{1} - s_{2} - is_{3}}{4}$$
(1)

where,  $s_n$  are complex spectra of the fully nonlinear surface elevation signals with 0,  $\pi/2$ ,  $\pi$  and  $3\pi/2$ phase shifts.  $S_0$  is the complex spectrum of the 2<sup>nd</sup> order difference components and  $S_1$ ,  $S_2$  and  $S_3$  are complex spectra of nonlinear super-harmonics for 1<sup>st</sup> (linear), 2<sup>nd</sup> (+) and 3<sup>rd</sup> harmonic, respectively.

New input amplitudes are then calculated based on the measured and the target amplitudes. In the same
 way input phases are also calculated

283  
$$a_{in}^{n}(f_{i}) = a_{in}^{n-1}(f_{i})a_{tgt}(f_{i}) / a_{out}^{n-1}(f_{i})$$
$$\phi_{in}^{n}(f_{i}) = \phi_{in}^{n-1}(f_{i}) + (\phi_{tgt}(f_{i}) - \phi_{out}^{n-1}(f_{i}))$$
(2)

where  $a_{in}{}^{n}(f_{i})$  and  $\phi_{in}{}^{n}(f_{i})$  are the amplitude and phase of an input spectral component at frequency  $f_{i}$ , respectively.  $a_{out}{}^{n}(f_{i})$  and  $\phi_{out}{}^{n}(f_{i})$  are the amplitude and phase of the corresponding spectral components of the measured/recorded output spectrum, respectively. The superscript *n* indicates the *n*-th iteration.  $a_{tet}(f_{i})$  and  $\phi_{tet}(f_{i})$  are set by the preselected target spectrum.

Iterations continue until the measured linearized amplitude spectrum matches the target amplitude spectrum, and the phases of the linearized part are zero at the desired location in the flume. By matching the measured amplitude spectrum to the target spectrum, NewWave-type focussed wave groups are generated in either physical or numerical wave flumes. The methodology has also been successfully applied to generate breaking waves by focussing in a CFD wave flume (Stagonas et al., 2018b) and in the present work it is applied to a CFD-based numerical model and a Lagrangian numerical flume with following and adverse sheared currents.

295

296 3. Numerical setup

Two approaches are used to replicate wave-current conditions generated in the physical flume, thus providing input conditions for the 3-D CFD model with the structure in place. In the first approach, the iteration scheme in physical experiments described in Section 2.2 is applied directly in the 2-D CFD model, while in the second approach, a Lagrangian model (Buldakov et al., 2015) is used to provide input conditions for the 3-D CFD model to reduce the size of the 3-D numerical CFD flume and shorten the simulation time.

In this section, we first present a general description of OpenFOAM-based numerical models and then the methodologies used for replicating the wave-current flow generated in the physical wave-flume are detailed. The accuracy and the efficiency of the methodologies are validated by comparing with the experimental measurements.

307

308 3.1 OpenFOAM-based numerical model

309 The CFD model based on OpenFOAM solves the Navier-Stokes (NS) equations or the Reynolds-

310 averaged Navier-Stokes (RANS) equations coupled with the continuity equation for the two-phase 311 combined flow of water and air with the incompressibility assumption,

$$\nabla \cdot \boldsymbol{u} = 0$$

$$\frac{\partial \rho \boldsymbol{u}}{\partial t} + \nabla \cdot (\rho \boldsymbol{u} \boldsymbol{u}) - \nabla \cdot (\mu \nabla \boldsymbol{u}) = -\nabla p^* - \boldsymbol{g} \cdot X \nabla \rho + \nabla \cdot (\rho \boldsymbol{\tau}) + \sigma \kappa \nabla \alpha$$
(4)

(3)

where  $\rho$  and  $\mu$  are the density and the dynamic viscosity of the mixed fluid, respectively, which are calculated following the equation (6) based on the Volume-of-Fluid (VOF) technique, which will be discussed below. u = (u, v, w) is the fluid velocity field in Cartesian coordinates, and  $p^*$  is the pressure in excess of hydrostatic pressure, defined as  $p^*=p - (g \cdot X)\rho$ . *g* is the acceleration due to gravity and X =(x, y, z) is the position vector. The stress tensor  $\tau$  is defined in a standard way (Jacobsen et al., 2012) and may include viscous and Reynolds stresses depending on solver settings.

- 320 Various turbulence closure models are implemented in OpenFOAM (e.g. Brown et al., 2016). However, the laminar flow model of OpenFOAM-2.4.0 is used in all computations reported here as both the 321 322 external wave fields and the wave force on the cylinder are dominated by inertial (potential flow) effects 323 (Chen et al., 2014; 2018). The reasonably good agreement between the numerical and experimental data shown in the following sections indicates that the consequences of viscosity and flow turbulence on the 324 free surface elevation and wave forces on the cylinder that are of interest in this study are negligible as 325 326 expected and supports the use of the laminar flow model. It is useful to note that turbulence modelling 327 may be important if drag forces and the formation of wakes are significant (Santo et al., 2015).
- The last term on the right-hand side of equation (4) is the effect of surface tension in which  $\sigma$  is the surface tension coefficient and  $\kappa$  is the curvature of the interface. The presence of surface tension is found to have minor effects in most civil engineering applications (Jacobsen et al., 2012; Larsen, 2018), thus,  $\sigma = 0$  is used in this study.
- The Volume-of-Fluid (VOF) technique is applied in OpenFOAM to locate and track the free surface (interface between air and water), with the following transport equation,

334 
$$\frac{\partial \alpha}{\partial t} + \nabla \cdot (\alpha u) + \nabla \cdot (\alpha (1 - \alpha) u_{\alpha}) = 0$$
 (5)

in which  $\alpha$  is the volume fraction function of water within each computational cell. This equation is similar to that proposed in Hirt and Nichols (1981), but with an additional compression technique (the last term on the left-hand side in which  $u_{\alpha}$  is an artificial compression term) to limit the numerical diffusion of the interface profile. The compression technique is developed by OpenCFD, and details can be found in Berberovic' et al. (2009).

340 The properties of the fluid at each cell are then calculated by weighting with the VOF function  $\alpha$ , which 341 ranges from 0 (if there is no traced fluid inside a cell) to 1 (when the cell is full of the traced fluid),

342 
$$\rho = \alpha \rho_{\text{water}} + (1 - \alpha) \rho_{air}; \ \mu = \alpha \mu_{\text{water}} + (1 - \alpha) \mu_{air} \tag{6}$$

The equations (3) – (5) are solved with the finite volume method in which the whole computational domain is discretized into a number of cells (Ferziger et al., 2002). The merged Pressure Implicit Splitting Operator (PISO) algorithm is then applied for each cell to decouple pressure from the momentum equation (Issa, 1986).

Both 3-D CFD models in this study use standard implementations of boundary conditions available at
 OpenFOAM (with the exception of inlet and outlet boundaries which will be discussed later). Detailed
 descriptions of available types of OpenFOAM boundary conditions are available in Greenshields (2015).

350 We are modelling waves on sheared current by disturbing the original parallel sheared flow, which is specified by the prescribed current profile. The profile results from the current boundary layer 351 developed on the solid bed and therefore originally satisfies no-slip conditions. Waves propagating over 352 353 the current perturb the flow and lead to the development of a secondary wave boundary layer near the bed. However, we consider a fast evolving transient wave. The wave boundary layer does not have time 354 355 to evolve and occupies only a very narrow region near the bed without affecting the rest of the flow. 356 Considerable number of additional mesh points are required to resolve this layer with no considerable impact on the overall wave behaviour. We therefore using a standard OpenFOAM free-slip condition 357

- 358 with zero gradient of tangential velocity as a bed/bottom condition. This allows smooth flow behaviour 359 near the wall without high mesh resolution. The same condition is applied on the side walls.
- 360 We use a no-slip boundary condition on the cylinder surface, and a computational mesh near the surface of the cylinder has the widths of cells ten times smaller than those away from the cylinder surface in 361 362 order to resolve the boundary layer. The wall-normal mesh size is selected to ensure that the dimensionless wall distance  $(y^+)$  is smaller than 5 based on the flat-plate boundary layer theory. The 363 364 mesh dimensions for the regions away from the cylinder are determined by convergence tests to ensure 365 that there are sufficient cells per wavelength to resolve propagating incident waves and wave-current-366 structure interactions; this will be discussed in more detail in the following section. It is found that 367 further refining the mesh inside the boundary layer has negligible influence on the flow-induced forces 368 on the cylinder. The Reynolds number  $Re (=\omega \eta_m^2/\nu)$  and the maximum local Keulegan-Carpenter 369 number KC (=2  $\omega \eta_m/D$ ) in this study are approximately 8.5 × 10<sup>4</sup> and 2.3, respectively.  $\omega$  is the peak 370 wave angular frequency, v is the kinematic viscosity and  $\eta_m$  is the maximum free surface elevation which is about 0.15 m in this study. The initial conditions and other boundary conditions follow the 371 372 same set-up as described in Chen et al. (2014) and Santo et al. (2017).
- The time step in OpenFOAM simulations is not fixed, but dynamically calculated to maintain a prescribed maximum Courant number  $Co = u\Delta t/\Delta x$  throughout the whole domain at all times.  $\Delta t$  is the time step,  $\Delta x$  is the cell size in the direction of the velocity and *u* is the magnitude of the velocity at that location (Courant et al., 1967). In this study an adjustable time step is used to achieve Co = 0.25, which is again determined by numerical experimentation, and not shown here for brevity. For details refer to Larsen et al. (2018).
- 379 OpenFOAM offers the extensive choice of numerical schemes and the iterative solvers/algorithm 380 settings for various terms in the equations (3) - (5) (Greenshields, 2015). These settings may have 381 significant effects on the performance of the CFD solvers in terms of accuracy and efficiency (Larsen 382 et al., 2018). The best choice can usually be determined from previous experience or on a case-by-case 383 basis by numerical experimentation. The scheme and solver choices used in this study are summarized 384 in Table A1 and detailed description of schemes, solvers and algorithms can be found in Greenshields 385 (2015). The combination of these choices in Table A1 has proved to work well and yield good results 386 when applied to nonlinear wave interactions with a vertical cylinder for ranges of flow conditions 387 studied in this work.
- 388
- 389 3.2 Direct application of the iterative wave generation methodology in CFD models
- This study uses and extends the toolbox 'waves2Foam' developed and released by Jacobsen et al. (2012) to realize wave generation and absorption in numerical wave flumes in OpenFOAM. The boundary conditions for generating waves are given analytically according to the linear wave theory, i.e. corresponding velocities and free surface elevations are specified at the input boundary faces. In this study, linear superposition of velocities of the spectral components of a wave group calculated using a desired spectrum (the spectrum of extracted linearized waves used here will be discussed later) is used to generate the focused wave group in the computational domain through a vertical wall.
- 397 A new boundary condition is developed within the framework of 'waves2Foam' to produce a vertically
- 398 sheared current. The sheared current profile is defined by a second-order polynomial which is obtained
- 399 by curve fitting the measured horizontal velocity profile at the model cylinder location. Figure 5
- 400 demonstrates the current profiles used in the CFD-based numerical simulations in this paper, and their
- 401 comparison with measured experimental profiles.
- The combined wave and current conditions are then generated by linearly superimposing the focussed wave group and sheared current at the inlet. The boundary condition for generating sheared current is also used at the outlet to ensure mass conservation.
- 405

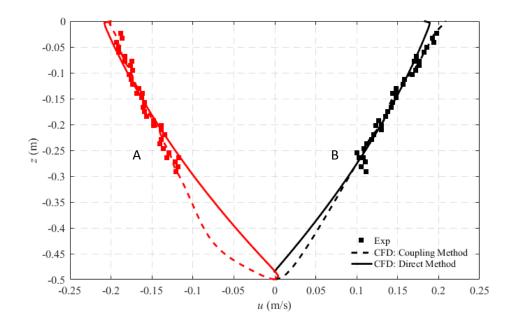


Figure 5 Comparisons of sheared current profile with depth obtained from the experiment and the numerical simulations at the location of the model cylinder for the cases with a sheared current and without waves. A -- Adverse current; B -- following current.

410

411 The wave-on-current focussing methodology described in Section 2.2 is now applied to generate 412 focussed waves on various flow conditions in the numerical wave flume. All iterations are performed 413 in a 2-D numerical flume replicating the physical flume at UCL. Although the target spectrum used in 414 the physical wave flume can be used as inputs for the first set of simulations, the linearized spectrum extracted from the actual experimental measurements is used instead to ensure a faster convergence to 415 416 the experimental measurements (i.e.  $a_{in}^{0} = a_{tgt} = a_{Linear}^{exp}$  and  $\phi_{in}^{0} = \phi_{tgt} = \phi_{Linear}^{exp}$  in equation 2). The free 417 surface elevations at x = -4.7 m (AMP in Figure 1) and x = 0 m (FP in Figure 1) in the numerical flume 418 are recorded and used for performing the amplitude and phase corrections following equations (1)-(2). 419 Generally, satisfactory/convergent results for all flow conditions considered in this work are obtained 420 within 1 or 2 iterations following the first set of simulations, i.e. in total three sets of 2-D simulations 421 are required. The final corrected set of boundary conditions is then used as input for the 3-D numerical model shown in Figure 6. Previously, the same approach has been successfully used to simulate extreme 422 423 forces induced by focussed waves on a following uniform current to a jacket structure, see Santo et al. 424 (2018).

The 3-D numerical flume (which is shown in Figure 6) consists of a rectangular domain with a vertical cylinder located at the centre of the flume. The total length of the flume is 13.7 m (~4 $\lambda_p$ ) with a distance of  $L_0$  between the inlet boundary and the vertical cylinder. The last 3 m (~ $\lambda_p$ ) of the numerical flume is occupied by the relaxation zone used to minimize wave reflections from the outlet.  $\lambda_p$  is the peak wavelength, which is ~3.2 m in this study. The width of the computational domain is 1.2 m, and the water depth *h* is 0.5 m, the same as those in the experiments. In the Direct Method,  $L_0 = 8.7$  m (~ 2.7  $\lambda_p$ ), the same as that in the experiments.

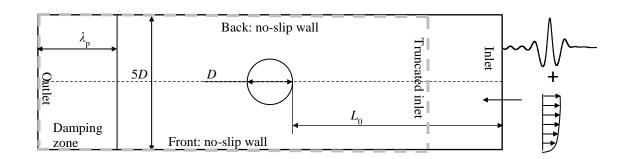


Figure 6 Layout of the computational domain. *D* is the diameter of the vertical cylinder and  $\lambda_p$  is the peak wavelength. The truncated inlet demonstrates the inlet boundary for the Coupling Method; the computational flumes for the Coupling and the Direct Methods are bounded by the dashed grey lines and the solid black lines, respectively.

438

The optimum set-up for the computational domain including its size and the mesh resolution is determined using numerical experimentation (not shown here for brevity but more details are given in Chen et al., 2014). The principle is to have the smallest possible domain size, thus minimum computational effort, while still maintaining the correct flow field around the structure.

443 Overall, the computational domain is divided into two areas, one with a coarser and one with a finer 444 mesh resolution. In particular, the area near the vertical cylinder and the layers near the air-water 445 interface are resolved with a finer mesh. Horizontal and vertical grid sizes for the coarser mesh are 446 about  $\lambda_p/240$  and  $H_p/12$ , respectively;  $\lambda_p$  and  $H_p$  are the peak wavelength and the peak wave height, 447 respectively. The cell size of the finer mesh is decreased by half, and cell sizes are graded so that the 448 size of the cells between the two areas varies smoothly.

449

## 450 Table 1 Parameters and computational costs used for two OpenFOAM-based models

Parameters	Direct Method Coupling Metho		
Overall length	13.7 (~4λp)	10 (~3λp)	
Overall width (	1.2 (~5D)	1.2 (~5D)	
Distance from inlet to c	8.7 (~2.7λp)	5 (~1.5λp)	
Distance from cylinder t	4.75 (~1.5λp)	4.75 (~1.5λp)	
Length of damping (relaxa	3 (~λp)	3 (~λp)	
Cell number (mil	~17.2	~12.6	
Maximum Courant nu	Maximum Courant number Co		
	Each 3-D	~12	~15.5
Computational costs (hrs)	Each 2-D	~ 1 (×12)*	
	Total	~ 24**	~15.5

451 \*In total 3 sets of 2-D simulations are required, and each set of 2-D simulations consists of 4 runs with successive

452 additional phase shifts of  $\pi/2$ , in total 12 2-D simulations are required, and each 2-D simulation requires ~1 hour 453 computational time.

454 \*\*Total time is calculated as the summary of the computational time required to calculate the corrected inlet 455 conditions using either the 2-D CFD model or the Lagrangian model and the time spent for the 3-D simulations.

456

The simulations were performed using the supercomputing facility at the Pawsey Supercomputing center which supports researchers in Western Australia. Utilising 48 cores for 3-D simulations, the

459 computational time is approximately 12 hours to obtain the results within the time scale of interest, i.e.

 $460 \sim 20$  s of modelled time corresponding to propagation of the wave group and its interaction with the

461 model structure. Each 2-D simulation used to calibrate the incoming wave group takes about 1 hour

using 24 cores. The geometric parameters used for 3-D simulations and computational costs requiredfor both 2-D and 3-D simulations are summarized in Table 1.

464

465 3.3 Generation of the incoming wave-current flow by coupling the Lagrangian and CFD models

The second method of generating wave-current conditions is based on reconstructing experimental surface elevation and kinematics of incoming waves on sheared currents by applying the iterative wave generation methodology (Section 2.2) to a Lagrangian numerical wave-current flume. The Lagrangian kinematics and the free surface elevation are then fed into a truncated numerical CFD wave flume with the cylinder present using an external forcing subroutine built onto waves2Foam and OpenFOAMbased numerical models.

472 A general Lagrangian formulation for two-dimensional flow of inviscid fluid with a free surface can be 473 found in Buldakov et al. (2006). We consider time evolution of coordinates of fluid particles x(a, c, t)474 and z(a, c, t) as functions of Lagrangian labels (a, c). The formulation includes the Lagrangian 475 continuity equation,

476 
$$\frac{\partial(x,z)}{\partial(a,c)} = J(a,c), \tag{7}$$

477 the Lagrangian form of vorticity conservation,

478 
$$\frac{\partial(x_i, x)}{\partial(a, c)} + \frac{\partial(z_i, z)}{\partial(a, c)} = \Omega(a, c)$$
(8)

479 and the dynamic free-surface condition,

480 
$$x_{tt}x_{a} + z_{tt}z_{a} + gz_{a}|_{c=0} = 0.$$
 (9)

Functions J(a, c) and  $\Omega(a, c)$  are given functions of Lagrangian coordinates and are defined by the initial conditions. J(a, c) is defined by initial positions of fluid particles associated with labels (a, c), and  $\Omega(a, c)$  is the vorticity distribution defined by the velocity field at t = 0. It is convenient to select initial undisturbed positions of fluid particles as Lagrangian labels  $(a, c) = (x_0, z_0)$ . This gives J = 1. For waves over a flat bed this defines a rectangular Lagrangian domain with c = 0 being the free surface and c = *h* being the bottom, where *h* is the undisturbed water depth. The boundary condition at the lower boundary can then be specified as,

488 
$$z(a, -h, t) = -h$$
 (10)

The presented Lagrangian formulation offers a simple treatment of vortical flows and therefore is suitable for modelling waves on vertically sheared currents. A sheared current can be defined by specifying vorticity depending only on the vertical Lagrangian coordinate *c*. For our choice of Lagrangian labels the parallel current can be specified as x = a + V(c)t; z = c, where  $V(c) = V(z_0)$  is the current profile. Substitution to (8) gives,

494 
$$\Omega(a,c) = \Omega(c) = V'(c) \tag{11}$$

Therefore, waves on a sheared current with an undisturbed profile  $V(z_0)$  are described by equations (7,8) with the free surface boundary condition (9), the bottom condition (10) and the vorticity distribution given by (11). Figure 5 demonstrates velocity profiles for adverse and following currents we are using in this paper and their comparison with measured experimental profiles. The current profiles applied for the Coupling Method (CFD: Coupling Method in Figure 5) are obtained from PIV and ADV (Acoustic Doppler Velocimetry) measurements of the current velocity using a Bezier smoothing algorithm. 502 For convenience and efficiency of numerical realisation, we modify the original problem (7-9) and write 503 it in the following form,

504 
$$\Delta_{t}\left(\frac{\partial(x,z)}{\partial(a,c)}\right) = 0; \ \Delta_{t}\left(\frac{\partial(x_{t},x)}{\partial(a,c)} + \frac{\partial(z_{t},x)}{\partial(a,c)}\right) = 0$$
(12)

505 and

506 
$$x_{tt}x_{a} + z_{tt}z_{a} + gz_{a} = RHS(a,t)|_{c=0}, \qquad (13)$$

507 where the operator  $\Delta_t$  denotes the change between time steps and the right-hand side of the dynamic 508 surface conditions includes various service terms. For calculations presented in this paper we use the 509 following additional terms,

510  

$$RHS = \left(\frac{1}{6}\delta_{a}^{2}x_{aa,t} - \frac{11}{12}g\delta_{i}^{2}z_{a,t}\right) - k(a)\left((x_{i} - V(c))x_{a} + z_{i}z_{a}\right) + P_{x}(a,t),$$
(14)

511 where  $\delta_a$  and  $\delta_t$  are the numerical mesh step in *a*-direction and the time discretization step. The first 512 term in (14) is the dispersion correction term, which increases the accuracy of the numerical dispersion 513 from second to fourth order. The second term enforces dissipation of surface perturbations. It is used 514 for absorbing reflections, and the dissipation strength is regulated by the coefficient k(a). The last term 515 in (14) is the prescribed time varying surface pressure gradient which is used for wave generation.

516 The numerical wave-current flume is created by specifying inlet and outlet boundary conditions, 517 distribution of surface dissipation k(a) and the surface pressure gradient  $P_x(a,t)$  providing free in- and 518 outflow of the current to and from the computational domain, generation of waves on/over the current 519 and absorption of waves reflected from domain boundaries.

The dissipation coefficient in the Lagrangian scheme is set to zero in the working section of the flume and gradually grows to a large value near the inlet and outlet boundaries. This results in a steady horizontal free surface at these boundaries which remain at their initial position z = 0 providing parallel inlet and outlet flows. This serves a double purpose. First, reflections from the boundaries are significantly reduced. Second, the boundary conditions at the inlet and outlet can be specified as the undisturbed velocity profile at the inlet and as a parallel flow at the outlet,

526 
$$x_t(a_{in},c,t) = V(c); z_a(a_{out},c,t) = 0.$$
 (15)

527 The wave is generated by creating an area in front of one of the wave absorbers where pressure 528 distribution of a prescribed shape is defined. Time-varying amplitude of this pressure disturbance is 529 used as a control input for wave generation. The problem is then solved numerically using a finite-530 difference technique. More details of the numerical method can be found in Buldakov (2013, 2014).

531 An additional difficulty with numerical realisation of the Lagrangian formulation on sheared currents is continuous deformation of the original physical domain. The accuracy of computations for strongly 532 533 deformed computational cells reduces considerably. In addition, parts of the deformed physical domain 534 can move outside the region of interest. To avoid these difficulties, we perform sheared deformation of the Lagrangian domain to compensate for the deformation of the physical domain. The deformation 535 takes place after several time steps and moves boundaries of the physical domain back to the original 536 537 vertical lines. After this Lagrangian labels are re-assigned to new values to preserve the rectangular 538 shape of the Lagrangian computational domain with vertical and horizontal lines of the computational 539 grid.

540 To reproduce experimental free surface elevation records, we use the iterative procedure described in 541 Section 2.2. Amplitudes and phases of spectral components of a pressure control signal are modified 542 iteratively to match amplitudes and phases of the calculated linearized surface elevation spectrum at 543 selected wave probes with target spectra. Linearized spectra of the actual experimental surface elevation

- 544 at locations x = -4.7 m (amplitude matching position) and x = 0 m (focus point) are used as targets for
- 545 the iterative procedure. Each numerical wave is generated with phase shifts of  $n\pi/2$ , with n = 0, 1, 2, 3.

This allows calculation of the linearized output signal of free surface elevation. The linearized output is then compared with the target, and corrections to the input spectrum for next iteration are calculated using the method described in Section 2.2. For further details of the iterative wave matching methodology refer to Buldakov et al. (2017). We apply the procedure to generate incoming waves for experimental cases presented in Section 2.1.

551 Lagrangian computations of the free surface elevation and flow kinematic time histories closer to the 552 structure are used as boundary conditions for a new, truncated 3-D numerical CFD wave flume (when 553 compared to the CFD domain of the Direct Method; dashed lines in Figure 6). The model cylinder is 554 centrally located in the new domain and the inlet is set at a distance of 5 m upstream from the cylinder. 555 Although Lagrangian calculations cover the full extent of the numerical flume, only the results at the inlet location (truncated inlet in Figure 6) are used in the 3-D CFD model using the Coupling Method. 556 The Lagrangian results are stored every 0.025 s (40 Hz) and are linearly interpolated to match the 557 internal time step of the CFD simulation. The same outlet relaxation zone (damping zone) used in the 558 559 Direct Method is used in the Coupling Method to minimise wave reflection and absorb outgoing mass 560 fluxes.

561 In contrast to the Direct Method, all iterations for the Coupling Method are conducted in the Lagrangian wave flume therefore allowing for a shorter CFD wave flume. The layout of the computational domain 562 is also shown in Figure 6. Compared with the Direct Method, the distance between the (truncated) inlet 563 564 boundary and the vertical cylinder is now 3 m smaller with  $L_0 = 5$  m, reducing total length of the numerical flume from 13.7 m to 10 m. More details about the CFD domains are summarised in Table 565 566 1, where it is also seen that the Coupling Method is in total (including the time required for the iterations) approximately 1.5 times faster than the Direct Method despite the fact that 3-D simulations with the 567 568 former method are found to require more computational time than simulations with the latter method. 569 This increase in computational time is attributed to the additional time required for the communications between the externally provided inlet boundary conditions and the OpenFOAM model. In particular, 570 small fluctuations in inlet boundary conditions require a smaller time step to ensure the stability of the 571 572 simulations.

573

### 574 3.4 Validation of wave-current generation methods

575 The computational results with both modelling approaches for wave-current interactions without the structure in place are now validated against experimental measurements. Free surface elevation time 576 577 histories at x = -4.7 m (amplitude matching position) and at x = 0 m (focus point) with following and 578 adverse sheared currents and without a current are presented in Figure 7. The outputs of the Lagrangian 579 numerical model are also included and are referred to as LaNM. An overall good agreement between experimental results and results from both the Direct and Coupling Methods is observed, with slightly 580 581 larger differences being found for the Direct Method. As discussed previously, wave-current generation 582 is different between the two numerical methods (Direct and Coupling Methods) and between numerical 583 methods and experiments, and thus the generation of different spurious waves is expected. This explains 584 the main differences between the methods and between calculations and experiments. The generation 585 of spurious long waves will be discussed in more detail in the following section.

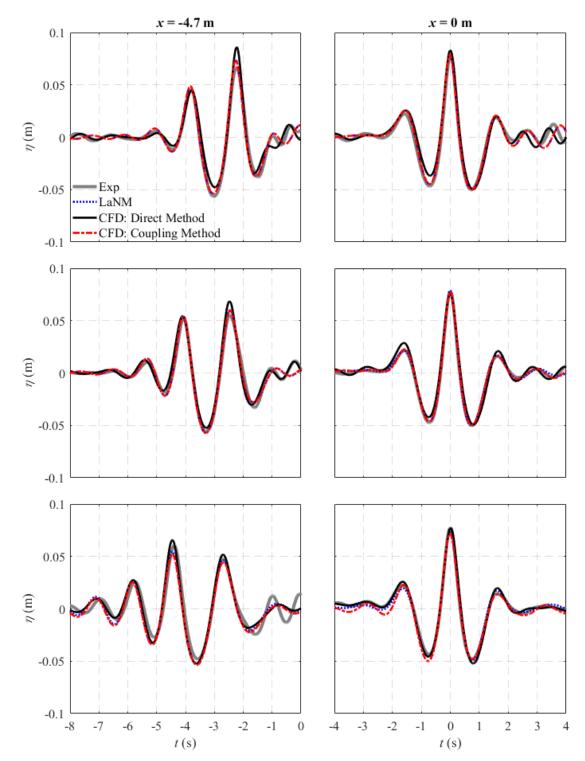


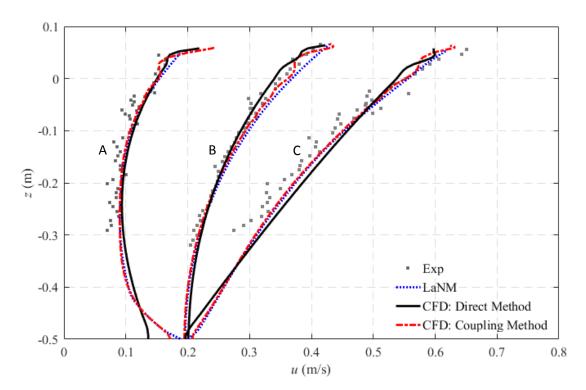
Figure 7 Comparisons of the free surface elevation time histories for cases with and without a sheared current. Left: x = -4.7 m (amplitude matching position); Right: x = 0 m (phase focus position). From top to bottom: following current, no current, and adverse current. All the results presented consider cases without the structure in place.

592

It is clear from Figure 7 that the wave shapes at the phase focussing position (right panels) for the cases with and without sheared currents are similar to each other as a result of the carefully controlled wave generation. The linearized spectrum at the focus point is the same for the experiments and computations for all the current cases considered (following, no-current and adverse current), while nonlinear contributions from higher order harmonics of the focussed wave group for different current cases are rather different. This leads to the differences in the main focused wave crests and the following crests at the focussed position. Further analysis on the harmonic structure of the free surface elevation will be presented in the following section. Additionally, it can be seen that the wave shapes at the amplitude matching position (left panels) are rather different from each other for all the current cases considered. This is due to the fact that the dispersion relations are different for waves on following, zero and adverse currents.

604 Flow kinematics computed at and below the peak of the main wave crest at focus are compared with PIV measurements in Figure 8. It can be seen from Figure 8 that the two CFD models (Direct and 605 Coupling Methods) and the Lagrangian model all provide very good predictions for the flow kinematics 606 below still water level (0 m). The largest discrepancies between the two CFD models (Direct and 607 Coupling Methods), the Lagrangian and experimental measurements are seen to occur in the vicinity of 608 609 wave crests. This difference in the velocity profile is partly caused by the inaccuracy of the numerical velocity profile (Figure 5). The limitation of the VOF method in reconstructing very steep and sharp 610 611 free-surfaces is also responsible for this difference around the interface (Wroniszewski et al., 2014). There is a sharp discontinuity of density at the interface, and the density-weighted velocity of air using 612 the VOF factor  $\alpha$  above the interface is close to zero, the velocity across the small interface between air 613

- 614 and water is smeared accordingly.
- 615



616

Figure 8 Velocity profiles under the wave crest for focussed wave groups for all three cases considered.
Numerical calculations and experimental measurements are included. A -- Adverse current; B -- no
current; C -- following current.

620

#### 621 4. Wave-current-structure interactions

622 Wave-current input conditions generated by the Direct Method and the Lagrangian model are now used 623 to simulate the wave-current-structure interaction using CFD-based models. We consider six cases, 624 including waves on following and adverse currents and without a current interacting with cylinders of 625 two diameters D = 0.25 m and D = 0.165 m

625 two diameters D = 0.25 m and D = 0.165 m.

626 Comparisons between computed and measured time histories of the horizontal load on the cylinder and 627 the free surface elevation at the front of the cylinder are presented in Figures 9 and 10. Results for 628 maximum free surface elevation and peak forces are summarized in Table A2 of Appendix 2.

629 Considering the cases with the larger cylinder (Figure 9), the time histories of the non-linear elevation
630 and horizontal force are predicted sufficiently well by either of the two approaches and the differences
631 are observed mostly in the amplitude of the first and the main crests which are also illustrated in Figure
632 8. The peak free surface elevation and horizontal force are generally very slightly under-predicted by

both approaches.

An equally good comparison between experimental and numerical results is reported in Figure 10 for the cases with the smaller cylinder. Differences in computed elevations are relatively larger than those for the larger cylinder, but differences in peak force predictions are as small as those for the larger cylinder. In all six cases considered the highest discrepancies between experimental and numerical force results are seen for the cases with adverse currents and in particular for the smaller cylinder. Computational results presented so far demonstrate a sufficient capacity of both CFD approaches (Direct and Coupling Methods) to model wave-current-structure interactions.

In the same time, CFD model cross-comparisons, by referring to the predicted elevation and force profiles, Figures 8 and 9, and the peak elevation and force, Table A2, show a good agreement and neither of the two approaches appears to be clearly superior to the other. Nevertheless, to further explore the source of the small differences observed between computations and between both computations and measurements, the fully non-linear elevation and force time histories are decomposed into their linear and non-linear components using the methodology described in Section 2.2.

647 The decomposed spectrum and the inverse Fourier transformation of each spectral part (e.g. time 648 histories of the  $2^{nd}$  order difference, linearized,  $2^{nd}$  order sum parts etc.) are shown in Figures 11-14.

649 The root mean square error for each spectral part is calculated as:

650 RMSE = 
$$\sqrt{\frac{\sum_{i=1}^{N} (a_{pi} - a_{mi})^2}{N}}$$
 (16)

where  $a_{pi}$  and  $a_{mi}$  are the spectral amplitudes of the i<sup>th</sup> (i = 0, 1, 2, 3) frequency predicted by the 651 computations, and measured in the experiments, respectively. N is the number of frequencies considered 652 in the calculations of the RMS error. N varies from 256 to 80, being larger for the linearized part and 653 decreasing for the nonlinear part. The range of frequencies considered for calculating the RMSE is 0 < 654  $f/f_p < 1$  for the 2<sup>nd</sup> order difference part ( $S_0$ ),  $0 < f/f_p < 3$  for the linearized part ( $S_1$ ),  $1 < f/f_p < 3$  for the 2<sup>nd</sup> order sum part ( $S_2$ ), and  $1.5 < f/f_p < 3.5$  for the third order part ( $S_3$ ). It is noted that each frequency 655 656 range was selected to include frequency components with non-negligible energy. As such, the integral 657 spectral error calculated with equation (16) is used as an integral measure to evaluate the level of 658 659 agreement between experimental and numerical results. The RMS errors for both methods and for all 660 test cases are shown in Figures 11 to 14 and they are summarized in Table A3 of the Appendix 2.

Considering the 2<sup>nd</sup> order difference harmonics, discrepancies are seen in the inverse Fourier time 661 662 histories of surface elevation and horizontal force on both cylinders. Given that waves are generated 663 linearly in the physical flume the occurrence of, e.g., the  $2^{nd}$  order spurious wave crests at approximately -3 s < t < 0 s in Figures 11 and 13 is not surprising. It is also worth noting that the same methodology 664 665 (Section 2.2) was used to reproduce the experimental results in the Coupling Method and the Direct Method. As such, the presence of spurious wave crests in the numerical results is also not surprising. 666 The Coupling Method is seen to somehow reproduce more closely 2<sup>nd</sup> order difference harmonics with 667 the experimental results, especially for the tests without currents. Given the variability in wave 668 generation methods between the flumes, and since the 2<sup>nd</sup> order wave generation is not employed, the 669 differences in the elevation of the 2<sup>nd</sup> order difference harmonic are expected. 670

The best agreement between experimental and numerical results is observed for the linearized part of the spectra. This is an expected outcome since with the iterative methodology the computations are forced to match the linearized part extracted from the experimental spectrum. However, it is illustrated by the time histories in Figures 11 to 14 and the RMS errors in Table A3, the Coupling Method is more

675 efficient in reproducing the experimental results.



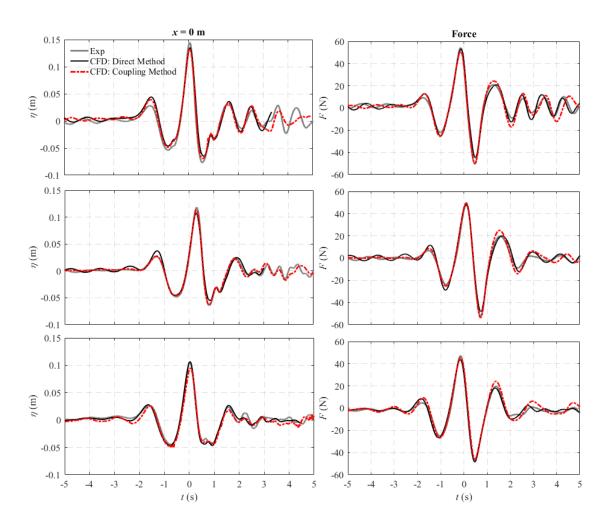


Figure 9 Comparisons of the free surface elevation time histories at the front of the cylinder (left) and the horizontal forces on the cylinder (right) for the larger cylinder (D = 0.25 m). From top to bottom: following current; no current; adverse current.

681

In contrast to spurious long waves (2<sup>nd</sup> order difference harmonic), spurious short-wave components (2<sup>nd</sup> order sum harmonic) travel with a celerity smaller than that of the wave group and thus they arrive at and interact with the structure after the focused wave. As a result, the agreement between experimental (elevation and force) measurements and computations for the 2<sup>nd</sup> order sum harmonics improves, see for example  $S_2$  for -1 s < t < 1 s in Figures 11 to 14. Particularly, for tests with the smaller cylinder, the RMS error for the forces predicted by the Coupling Method is smaller but once again the difference with the errors calculated for the Direct Method is not significant.

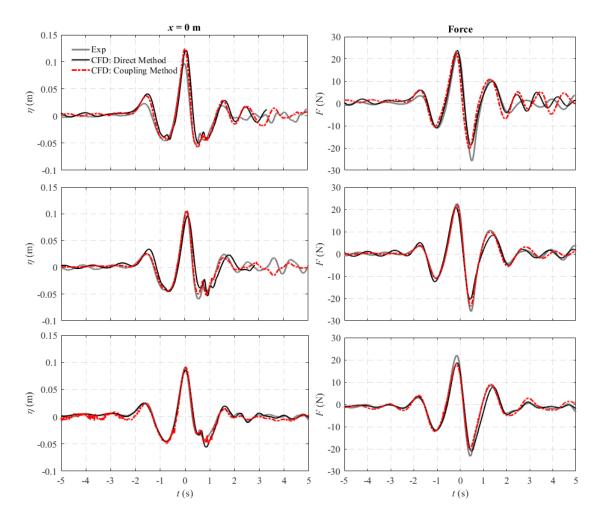


Figure 10 Comparisons of the free surface elevation time histories at the front of the cylinder (left) and the horizontal forces on the cylinder (right) for the smaller cylinder (D = 0.165 m). From top to bottom: following current; no current; adverse current.

694

Similar conclusions are drawn from Figures 11 to 14 and Table A3 about the 3<sup>rd</sup> order sum harmonics 695 albeit the agreement between the 3<sup>rd</sup> order horizontal forces is not as impressive as the agreement 696 between experimental and numerical free surface elevation. Although the combination of four phase 697 shifted elevation/force signals is sufficient to efficiently isolate the 3rd order harmonics (Buldakov et al., 698 699 2017), the very small amplitude of the 3<sup>rd</sup> order harmonics challenges the accuracy limits of experimental measurements. With this in mind, the performance of both Methods is considered to be 700 satisfactory with the Coupling Method results being slightly closer to the experiments. Despite the small 701 702 amplitudes, the 3<sup>rd</sup> order force harmonics are still important since they are often related to the 'ringing' 703 phenomenon.

704 With regards to the inter-comparison of the two numerical approaches, Figures 11 to 14 reveal no 705 significant differences and neither model is seen to outperform the other. The small differences in the 706 performance of the Coupling and the Direct Method are likely due to the fact that the Lagrangian model 707 reconstructs the experimental input conditions with slightly higher precision; see also Figure 7. Small 708 discrepancies in the 2<sup>nd</sup> order difference components can be attributed to the different wave generation 709 methods adopted, but they are not seen to result in significant discrepancies in the overall computations 710 of free surface elevation and force time series, e.g. Figures 9 and 10. In general, RMS errors for the 711 Coupling Method tend to be smaller than those for the Direct Method. This in combination with the 712 smaller computational effort required (Table 1) shows an advantage in favour of the Coupling Method.

#### 714 5. Force decomposition

We have demonstrated in Figures 12 and 14 that the harmonic structure of forces on a cylinder in waves and sheared currents can be accurately decomposed into harmonic contributions using the four-phase based decomposition method in Section 2.2. Chen et al. (2018) showed that the harmonic structure of force on a vertical cylinder in a wave group without current can be adequately modelled based on only the linear component as follows. We write the linear component in time as

$$F_1 = \mathcal{F}_1 f_1$$

where  $\mathcal{F}_1$  is the peak of the envelope of  $F_1$  in time and  $f_1$  carries all the phase information and group structure in time. Then the assumed form of the total force in time is

$$\frac{F}{\rho g R^{3}} = \frac{\mathcal{F}_{1}}{\rho g R^{3}} [f_{1}] + S_{FF2} (\frac{\mathcal{F}_{1}}{\rho g R^{3}})^{2} [\alpha_{FF2} (f_{1}^{2} - f_{1H}^{2}) + \beta_{FF2} (2f_{1}f_{1H})] + S_{FF3} (\frac{\mathcal{F}_{1}}{\rho g R^{3}})^{3} [\alpha_{FF3} f_{1} (f_{1}^{2} - 2f_{1H}^{2}) + \beta_{FF3} (f_{1H} (3f_{1}^{2} - f_{1H}^{2}))] + S_{FF4} (\frac{\mathcal{F}_{1}}{\rho g R^{3}})^{4} [\alpha_{FF4} (f_{1}^{4} - 6f_{1}^{2}f_{1H}^{2} + f_{1H}^{4}) + \beta_{FF4} (4f_{1}f_{1H} (f_{1}^{2} - f_{1H}^{2}))] + \dots$$
(18)

(17)

723

The force approximation contains Stokes-like amplitude terms 
$$\mathcal{F}_1 / \rho g R^3$$
 based on the peak amplitude  
of the linear force component, non-dimensional force coefficients at each order  $S_{FFn}$  and phase  
coefficients  $(\alpha_{FFn}, \beta_{FFn})$  with  $\alpha_{FFn}^2 + \beta_{FFn}^2 = 1$ .  $R = D/2$  is the radius of the cylinder. The subscript  $H$   
denotes the Hilbert transform of the  $f_1$  function in time, and the increasingly complicated products of  $f_1$   
and  $f_{1H}$  denote the shape of the *n*th harmonic in time. The coefficients  $S_{FFn}$  and  $\alpha_{FFn}$ ,  $\beta_{FFn}$  are estimated  
by weighted fits, as described in Chen et al. (2018). Chen et al. (2018) showed that this approximate  
form works well for all the harmonics up to the 5<sup>th</sup> but that the 3<sup>rd</sup> harmonic fits are less good.

Here we briefly demonstrate that these decompositions work equally well for forces from waves on sheared currents, and that the form of the current affects the force coefficients  $S_{FFn}$  significantly, but the phase terms ( $\alpha_{FFn}$ ,  $\beta_{FFn}$ ) only slightly. The coefficient values are given in Table A4 of Appendix 2. We note in passing that the assumed form of the inline force on the cylinder (equation 18) neglects drag completely. Clearly, for the flow conditions reported here, unsteady inviscid components dominate the force time histories.

The reconstructed harmonics up to the 4<sup>th</sup> harmonic are compared to the extracted experimental 737 harmonics in Figure 15 for the larger cylinder, and in Figure 16 for the smaller cylinder. The 738 experimental harmonics are extracted with the four phase decomposition method of Section 2.2 and 739 Fitzgerald et al. (2016) and the 4<sup>th</sup> sum harmonic is separated from the 2<sup>nd</sup> order difference term by 740 digital filtering. It can be seen from the figures that the reconstructions of the 2<sup>nd</sup> and 4<sup>th</sup> harmonics 741 742 work well, and for both cylinders the amplitudes of the harmonics are largest for the following current and smallest for the adverse current. These bracket the case with no current. The 3<sup>rd</sup> harmonic 743 contributions are fitted less well with significant structure outside the time range of the (linear 744 745  $envelope)^3$  as discussed by Chen et al. (2018) for cases without current. That is, obvious wiggles outside the envelopes of 3<sup>rd</sup> harmonics are observed as shown in Figures 15 and 16; the envelopes of 3<sup>rd</sup> 746 747 harmonics are approximated by raising the linear envelope to the power three, and then scaled to fit the measured envelopes of the 3<sup>rd</sup> harmonic component by a least-squares method (Chen et al., 2018). 748 749 Further analysis of the forces and scattered waves is left for a follow-on paper.

750

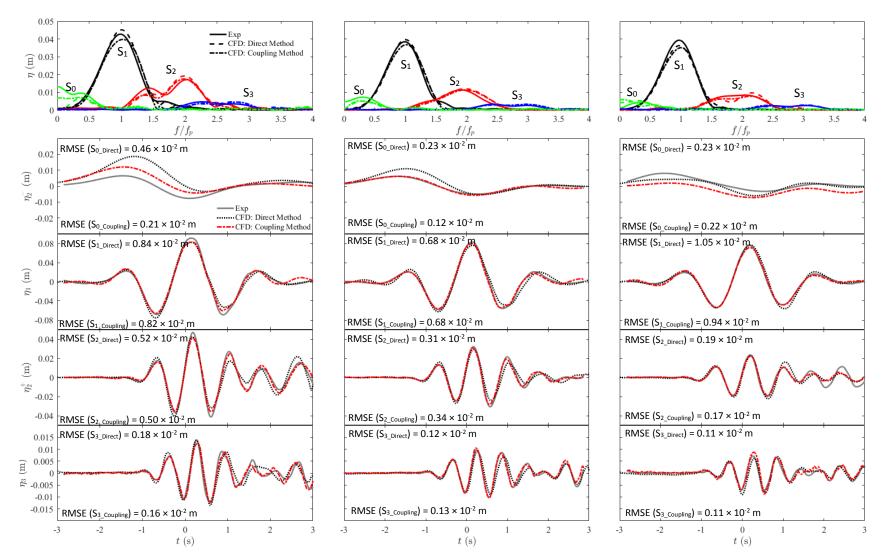
#### 751 6. Conclusions

Two approaches are proposed and used in this numerical study to generate nonlinear focussed wave groups propagating on a sheared current so as to allow an investigation of complex interactions between a combined wave-current flow and a vertical surface piercing cylinder, with applications to problems

- 755 in coastal engineering. Both approaches employ an iterative wave-on-current focussing methodology to ensure controlled wave-current generation. In the first approach, i.e. the Direct Method, the iterative 756 757 methodology is applied directly in a 2-D OpenFOAM model to provide input conditions for a 3-D OpenFOAM model, while in the second approach, i.e. the Coupling Method, the input wave-current 758 759 kinematics of the 3-D OpenFOAM model is created in a faster numerical model. In this study, a 760 Lagrangian numerical wave-current flume is used as the fast model for reconstructing experimental 761 surface elevation and kinematics of incoming focussed waves on sheared currents. There is no necessity 762 to have such a long distance between the wavemaker and the structure to ensure a full development of 763 the combined wave-current flow before the complex interactions with the structure. Thus using the 764 Coupling Method allows a smaller 3-D computational domain and shorter simulation time for modeling 765 wave-current-structure interactions when compared to the Direct Method.
- 766 It is worth noting that the wave-on-current focussing methodology applied in this study considers only 767 the linearized part of wave group spectrum, and phase and amplitude corrections are performed at 768 different locations to improve the effectiveness and convergence of the iterative procedure; the phases 769 are corrected at the pre-selected focus location, and amplitudes are corrected at a location well before 770 the focus position.
- 771 Good agreement between the experimental and numerical results demonstrates that both numerical 772 methods are capable of replicating experimental wave-current flows, and then accurately modelling 773 interactions between surface piercing cylinders and focussing waves on sheared currents. It is found 774 that the Coupling Method is computational cheaper due to the application of the iterative wave-oncurrent focusing methodology in the faster Lagrangian model. More specifically, for the simulations 775 776 considered in this study the computational efficiency is increased by a factor of approximately 1.5. 777 Overall, both approaches can be recommended as practical methods for studies of wave-current 778 interactions with structures, especially the Coupling Method that has a higher computational efficiency. 779 It is worth mentioning that the Lagrangian model can be coupled with various models and solvers, and 780 is thus applicable for a wide range of wave-current-structure interaction problems.
- 781 It is also found that the Stokes-wave perturbation expansion of Chen et al. (2018) can be generalized to 782 cylinder loads arising from wave groups on adverse and following currents and without a current. The 783 higher-order harmonic shapes can be estimated from knowledge of the linear components alone, and 784 the actual time history at each harmonic can be reconstructed to a reasonable approximation from the 785 linear component time history, using an amplitude coefficient and a phase angle at each harmonic. The  $2^{nd}$  and  $4^{th}$  harmonic force coefficients are found to be the largest on a following current, and the smallest 786 on an adverse current. The results for waves without a current sit in between. The 3<sup>rd</sup> harmonic forces 787 788 fit the simple expansion less well, as observed by Chen et al. (2018) for the case of no current. The 789 application of this reconstruction method to a wide range of wave-current conditions will be considered 790 in future work.
- 791
- 792 Acknowledgement

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798 Performance Computing (HPC) facility at the University of Bath in carrying out parts of this work.



799

Figure 11 Harmonic components of the free surface elevation at the front face of the larger cylinder (D = 0.25 m). From top to bottom: Amplitude spectra of the free surface elevation, 2<sup>nd</sup> order difference harmonic, linear harmonic, 2<sup>nd</sup> order sum harmonic, and 3<sup>rd</sup> harmonic. From left to right: following current; no current; adverse current.

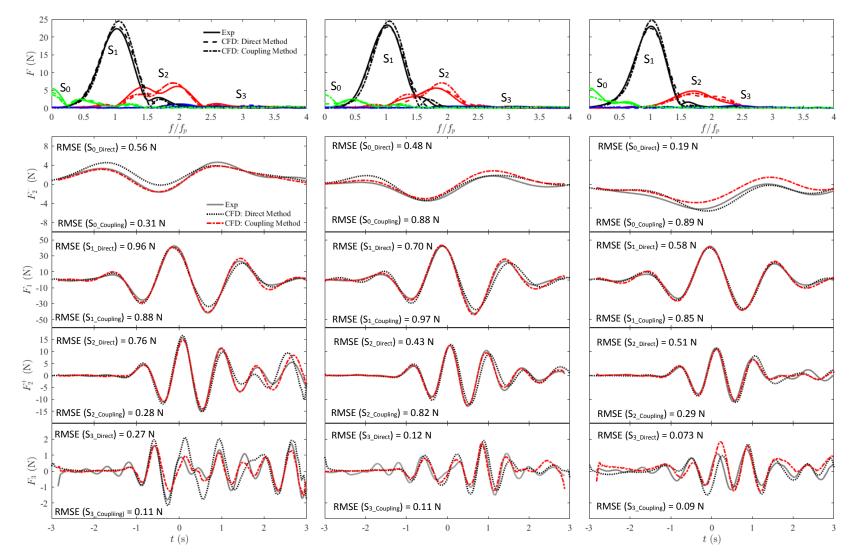
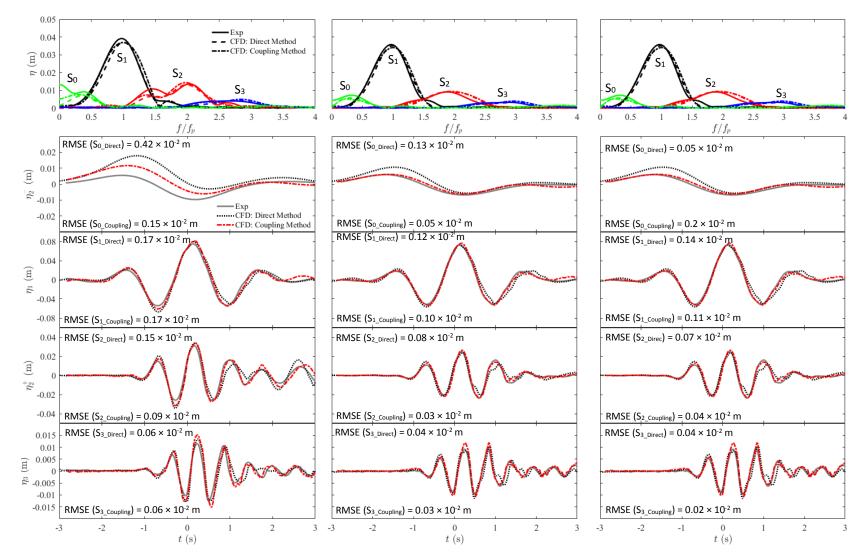
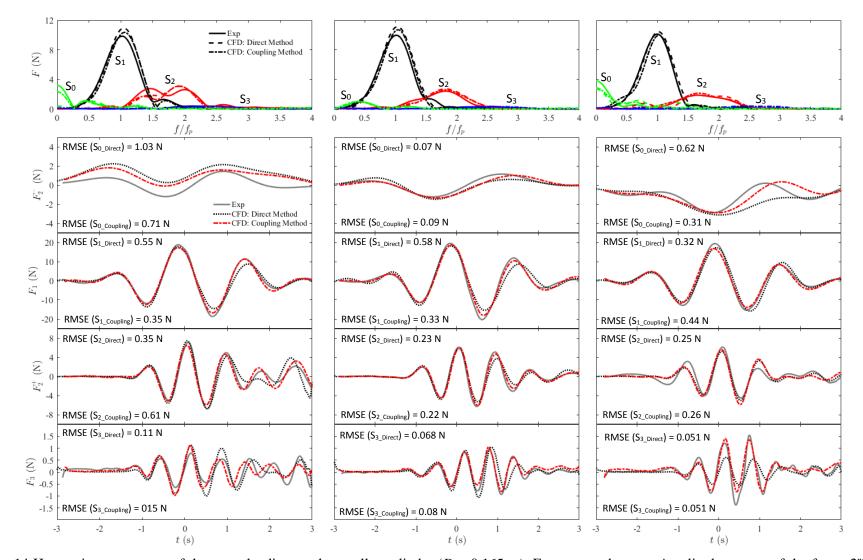


Figure 12 Harmonic components of the wave loading on the larger cylinder (D = 0.25 m). From top to bottom: Amplitude spectra of the force, 2<sup>nd</sup> order difference harmonic, linear harmonic, 2<sup>nd</sup> order sum harmonic, and 3<sup>rd</sup> order harmonic. From left to right: following current; no current; adverse current.



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Figure 13 Harmonic components of the free surface elevation at the front face of the smaller cylinder (D = 0.165 m). From top to bottom: Amplitude spectra of the free surface elevation, 2<sup>nd</sup> order difference harmonic, linear harmonic, 2<sup>nd</sup> order sum harmonic, and 3<sup>rd</sup> order harmonic. From left to right: following current; no current; adverse current.



810

Figure 14 Harmonic components of the wave loading on the smaller cylinder (D = 0.165 m). From top to bottom: Amplitude spectra of the force, 2<sup>nd</sup> order difference harmonic, linear harmonic, 2<sup>nd</sup> order sum harmonic, and 3<sup>rd</sup> order harmonic. From left to right: following current; no current; adverse current.

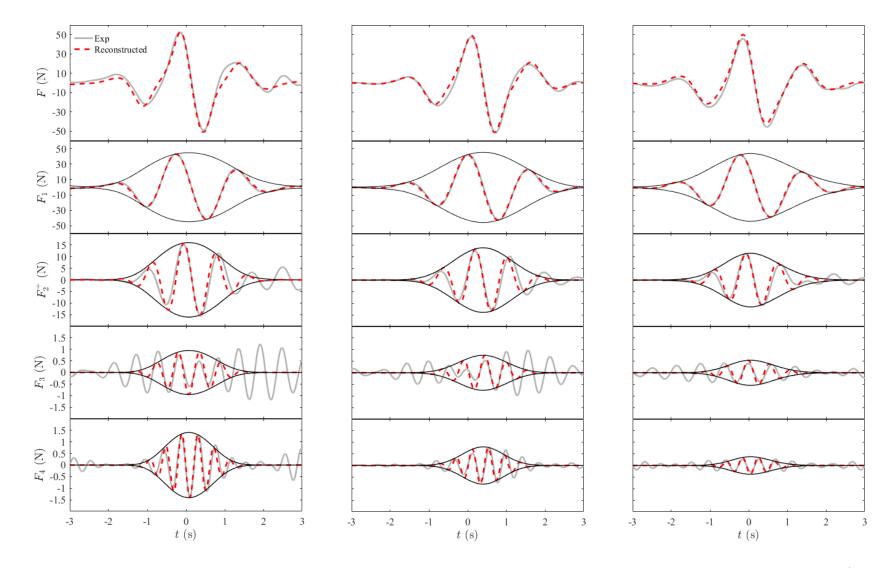




Figure 15 The reconstruction of horizontal wave loading on the larger cylinder (D = 0.25 m). From top to bottom: Total force, linear harmonic, 2<sup>nd</sup> order sum harmonic, 3<sup>rd</sup> order harmonic, and 4<sup>th</sup> order harmonic. From left to right: following current; no current; adverse current.

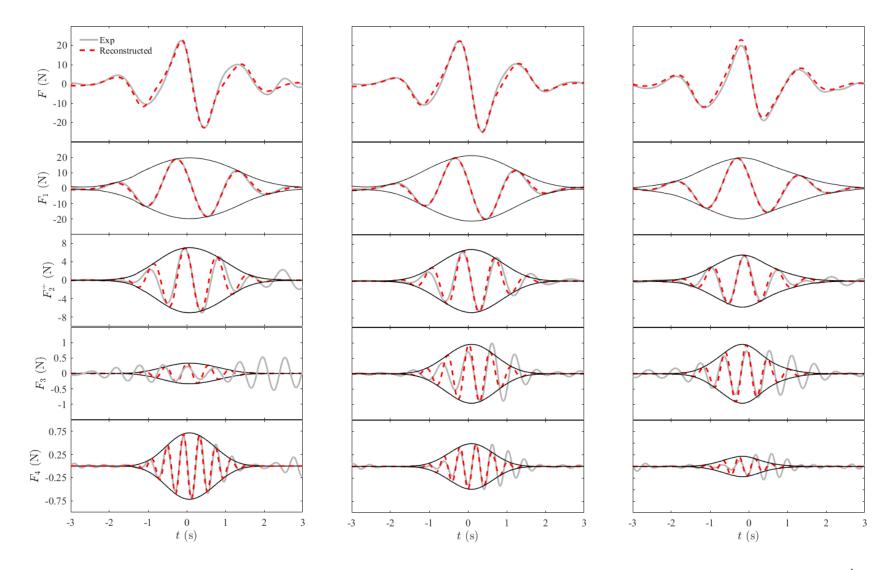




Figure 16 The reconstruction of horizontal wave loading on the smaller cylinder (D = 0.165 m). From top to bottom: Total force, linear harmonic, 2<sup>nd</sup> order sum harmonic, 3<sup>rd</sup> order harmonic and 4<sup>th</sup> order harmonic. From left to right: following current; no current; adverse current.

819 Appendix 1. Numerical schemes and solvers

820 The selected numerical schemes used to discretize the different terms in the governing equations, and

the settings for the linear solvers and for the solution algorithm are summarized in Table A1. In

822 OpenFOAM, they can be specified in the *fvSchemes* and *fvSolution* files, respectively.

823 The treatment of the first order time derivative terms  $(\partial/\partial t)$  in the momentum equations is specified in 824 the *ddt* scheme. Three transient schemes are widely used for engineering applications including *Euler*, 825 Backwards and CrankNicolson (CN). The Euler scheme corresponds to the first-order forward Euler scheme, while Backwards is a second-order implicit time discretization scheme in which the results 826 827 from the current and two previous time steps are used. A blending factor is introduced in the 828 CrankNicolson (CN) scheme to improve its stability and robustness; the blending factor of 1 829 corresponds to a pure CN scheme with a second-order accuracy, and 0 corresponds to pure Euler. The simulations with the *Euler* scheme are faster but may lead to a heavy diffusion of the air-water interface. 830 831 The use of a CN scheme is recommended for waves with long propagation distances and times (Larsen 832 et al., 2018).

833 One of major challenges in CFD calculations is the treatment of convective/advective terms in the 834 governing equations. Different schemes are specified for different convective terms as they are 835 fundamentally different. The standard finite volume discretization of Gaussian integration is implemented in OpenFOAM in which the integral over a control volume is converted to a surface 836 837 integral using the Gauss theorem. Accordingly, the word "Gauss" is specified in the numerical schemes. 838 The Gaussian integration requires the interpolation of the field variable from cell centres to face centres using for example central/linear or upwind differencing. The former is second-order accurate, but may 839 840 cause oscillations (unboundedness) in the solution, while the latter is first order accurate, thus, is more 841 diffusive. In lieu of this, various total variation diminishing (TVD) and normalized variable diagram 842 (NVD) schemes that utilize combined upwind and linear differencing are implemented in OpenFOAM, 843 including schemes of *limitedLinear* and *vanLeer*. The use of upwind differencing or linear upwind differencing for the momentum flux is preferable if the loads on the structure are of main concern, such 844 845 as the cases in this study. A similar conclusion is presented in Larsen et al. (2018).

Generally, the linear schemes are used for calculating the gradients and the interpolation from cell centres to face centres although higher order accurate schemes are available. The *laplacian* scheme requires the specification of an interpolation scheme for e.g. the dynamic viscosity  $\mu$ , and a surface normal gradient scheme for e.g.  $\forall u$ . Again, linear schemes are often used with orthogonality corrections for surface normal gradients. For more detailed descriptions on various numerical schemes in OpenFOAM, the reader is referred to the OpenFOAM user's guide (Greenshields, 2015) and programmer's guide (Greenshields, 2015) as well as Larsen et al. (2018).

853 The iterative solvers, solution tolerances and algorithm settings for solving the discretised algebraic 854 equations are specified in the *fvSolution* file. Various iterative solvers are implemented in OpenFOAM, 855 including preconditioned (bi-) conjugate gradient solvers (PCG/PBiCG) and smoothSolver in which the 856 specification of preconditioning of matrices (preconditioner) and smoother is required, respectively. The generalised geometric-algebraic multi-grid (GAMG) solver is also commonly used in which the 857 initial guess of the accurate solution on the finer simulation mesh is obtained by mapping the quicker 858 859 solutions on a coarser mesh to this finer mesh. Generally, the GAMG solver is quicker than the 860 smoothSolver, whereas the latter may yield more accurate results. The use of PCG/PBiCG solver sits 861 in between. Detailed descriptions refer to the OpenFOAM user's and programmer's guides (2015).

862 In this study, the compression velocity  $u_{\alpha}$  in the equation (5) equals to the flow velocity at the interface 863 by specifying *cAlpha* to be 1. A larger value of *cAlpha* leads to a sharper interface but also the 864 appearance of wiggles in the air-water interface which is found to be responsible for un-physical 865 steepening of waves and over-estimations of wave celerity (Larsen et al., 2018). Whereas, the use of a

- smaller *cAlpha* reduces the wiggles but at the same time leads to a more significant smearing interface.
- 867 Another two important controls over the  $\alpha$  equation are *nAlphaCorr* and *nAlphaSubCycles*; the former 868 specifies how many times the  $\alpha$  field should be solved within a time step, and the latter represents the 869 number of sub-cycles for the  $\alpha$  equation within a given time step.
- 870 As aforementioned, the PISO algorithm is applied in this study, thus, nOuterCorrectors = 1, and the
- 871 parameter *nCorrectors* is the number of pressure corrector iterations in the PISO loop and the 872 *momentumPredictor* is a switch that controls solving of the momentum predictor. Each time step will
- be begun by solving the momentum equation rather than the pressure equation if the momentum
- 873 be begun by solving the momentum equation rather th 874 predictor is turned on.
  - 875

Numerical schemes								
Terms in equations	Representation in OpenFOAM	Discretization schemes	Description					
Time derivatives	ddt	Euler	First order forward Euler scheme					
Gradients	grad	Gauss linear						
Divergence (momentum flux)	div(rho*phi, U)	Gauss linearUpwind, grad(U)	Second order, upwind- biased, specification of velocity gradient is required.					
Divergence (mass flux)	div(phi, alpha)	Gauss vanLeer	Total variation diminishing (TVD)					
Divergence	div (phib, alpha)	Gauss linear						
Laplacian	laplacian	Gauss linear corrected	Interpolation and snGrad schemes are required.					
Interpolation	interpolation	linear						
Surface normal gradient	snGrad	Linear with orthogonality correction						
Iterative solvers								
Equations	Variable field	Solvers						
Pressure <i>p</i> *	pcorr/p_rgh/ p_rghFinal							
Velocity U	U	smoothSolvers, symGaussSeidel, 1e-06, 0	tolerance, relative					
VOF function $\alpha$	alpha.water	tolerance						
		Algorithm controls						
Artificial compression term $u_{\alpha}$	cAlpha	1	$u_{\alpha} = u$ in which $u$ is the flow velocity at the interface					
PISO loop	SO loop momentumPredictor no		Loop starts by solving the pressure equation					
PIMPLE loop	nOuterCorrectors	1	PISO is used, otherwise, PIMPLE is used.					
PISO loop	nCorrectors	3	pressure corrector iterations					
Loop over the	Loop over the nAlphaCorr 2		$\alpha$ corrector iterations					
$\alpha$ equation	nAlphaSubCycles	1	Number of sub-cycles					

6 Table A1 The selected numerical schemes and iterative solvers.

877 Appendix 2. Detailed model comparisons and coefficients used for the reconstruction

Table A2 summarized the results for maximum surface elevation and peak forces, and all differences

in Table A2 are calculated with respect to the experimental data and they are only used for a qualitative

880 model comparison. Integral spectra errors are reported in Table A3 and used as an approach

- demonstrating model accuracy in depth. Table A4 summarized the coefficients used for reconstructing
- the higher order harmonics from the linear components alone, as shown in Figures 15-16.
- 883

Table A2 Comparisons between the two models in terms of wave crests/troughs and peak forces

Cases				Direct	t Method	Coupling Method	
Cylinders	Parameters	Current (Heading)	Exp.	Num.	Differences (%)	Num.	Differences (%)
	Wave crest	Following	0.144	0.135	-6	0.135	-6
	(m)	No current	0.118	0.108	-8	0.114	-3
		Adverse	0.107	0.107	0	0.093	-13
	Wave	Following	-0.077	-0.065	-16	-0.070	-9
	trough (m)	No current	-0.064	-0.055	-14	-0.061	-5
D =		Adverse	-0.042	-0.046	10	-0.043	2
0.25 m	Positive	Following	54.17	53.07	-2	52.67	-3
	peak forces (N)	No current	48.10	48.10	0	49.70	3
	(11)	Adverse	46.90	43.89	-6	45.46	-3
	Negative peak forces (N)	Following	-49.04	-44.75	-9	-48.90	0
		No current	-51.25	-48.07	-6	-53.63	5
		Adverse	-46.03	-48.33	5	-44.94	-2
	Wave Crest	Following	0.100	0.120	20	0.120	20
	(m)	No current	0.105	0.096	-9	0.105	0
		Adverse	0.090	0.085	-6	0.091	1
	Wave	Following	-0.053	-0.049	-8	-0.055	4
	trough (m)	No current	-0.059	-0.054	-8	-0.051	-14
D = 0.165	e v	Adverse	-0.040	-0.055	38	-0.047	18
m	Positive peak forces (N)	Following	22.51	23.84	6	22.51	0
		No current	22.51	20.93	-7	21.69	-4
		Adverse	22.00	18.28	-17	18.28	-17
	Negative	Following	-24.53	-18.67	-24	-20.17	-18
	peak forces	No current	-25.71	-19.66	-24	-22.64	-12
	(N)	Adverse	-23.04	-20.9	-9	-19.90	-14

Cases			Direct Method				Coupling Method			
Cylinders	Parameters	Current (Heading)	2 <sup>nd</sup> order dif.	Linea rized	2 <sup>nd</sup> order sum	3 <sup>rd</sup> order	2 <sup>nd</sup> order dif.	lineariz ed	2 <sup>nd</sup> order sum	3 <sup>rd</sup> order
	Free surface elevations $(\times 10^{-2} \text{ m})$	Following	0.46	0.84	0.52	0.18	0.21	0.82	0.50	0.16
		No current	0.23	0.68	0.31	0.12	0.12	0.68	0.34	0.13
D =		Adverse	0.23	1.05	0.19	0.11	0.22	0.94	0.17	0.11
0.25 m	Forces (N)	Following	0.56	0.96	0.76	0.27	0.31	0.88	0.28	0.11
		No current	0.48	0.70	0.43	0.12	0.88	0.97	0.82	0.11
		Adverse	0.19	0.58	0.51	0.073	0.89	0.85	0.29	0.09
	Free surface elevations (× 10 <sup>-2</sup> m)	Following	0.42	0.17	0.15	0.06	0.15	0.17	0.09	0.06
		No current	0.13	0.12	0.08	0.04	0.05	0.10	0.03	0.03
D =		Adverse	0.05	0.14	0.07	0.04	0.2	0.11	0.04	0.02
0.165 m	Forces (N)	Following	1.03	0.55	0.35	0.11	0.71	0.35	0.61	0.15
		No current	0.07	0.58	0.23	0.068	0.09	0.33	0.22	0.08
		Adverse	0.62	0.32	0.25	0.051	0.31	0.44	0.26	0.051

# 886Table A3 Root-mean-square errors for various harmonics in spectra space

kR	1-1-		Orden	Coefficients			
ĸĸ	kh		Order	Following	No current	Adverse	
		$\begin{array}{c} \text{Amplitude} \\ (S_{FFn}) \end{array}$	2	3.03	2.58	2.06	
0.242	0.97		3	0.34	0.27	0.18	
0.242			4	0.98	0.54	0.23	
(larger		Phase (deg.) $(\alpha_{FFn}, \beta_{FFn})^*$	2	97	94	73	
cylinder)			3	49	305	148	
			4	145	123	68	
	0.07	Amplitude (S <sub>FFn</sub> )	2	2.01	1.89	1.51	
0.1.00			3	0.12	0.34	0.32	
0.160			4	0.33	0.22	0.09	
(smaller	0.97	Phase (deg.) $(\alpha_{FFn}, \beta_{FFn})^*$	2	99	99	81	
cylinder)			3	183	245	190	
			4	165	128	48	

# 888 Table A4 Coefficients for reconstructing the higher order harmonics for all three flow conditions

889

\* Phase =  $\arctan(\beta_{FFn} / \alpha_{FFn})$ 

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